

C.V.RAMAN POLYTECHNIC, BHUBANESWAR



C.V.Raman Polytechnic

Quality Education for the New Millennium

**LECTURE NOTE
STRUCTURAL DESIGN-II,(Th.2)**

SEM- 5TH

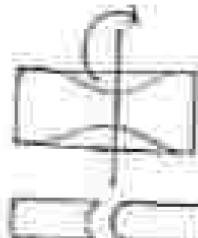
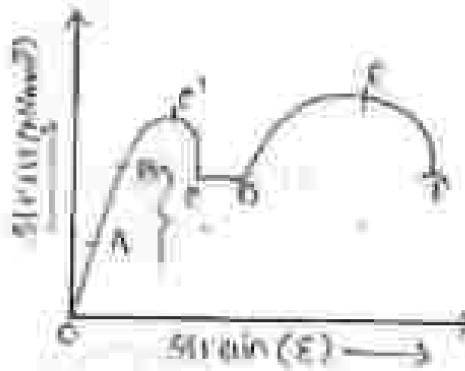
BRANCH- CIVIL ENGINEERING

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Internal forces are axial force, shear force, bending moment, tension



Ductile failure

A : Proportional limit

B : Elastic limit

C' : Upper yield point

C : Lower yield point

E : Ultimate strength point/stress corresponding to ultimate load.

F : Breaking stress corresponding to breaking load.

OAB : Plastic region

CD : Plastic yielding region

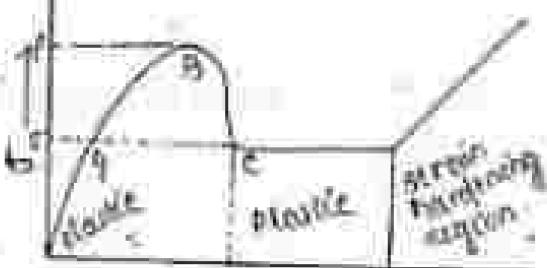
EF : Strain softening region

DE : Strain hardening region

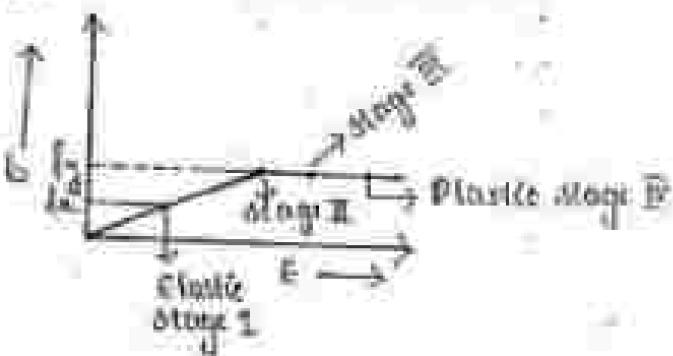
Strain increases but stress still will ultimate load is reached

Answer:-

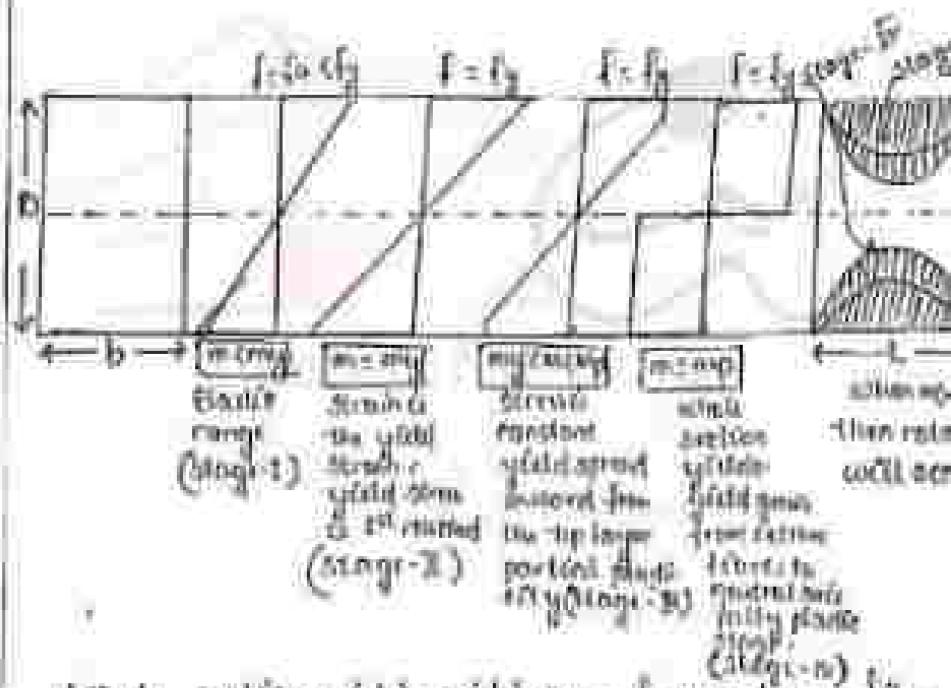
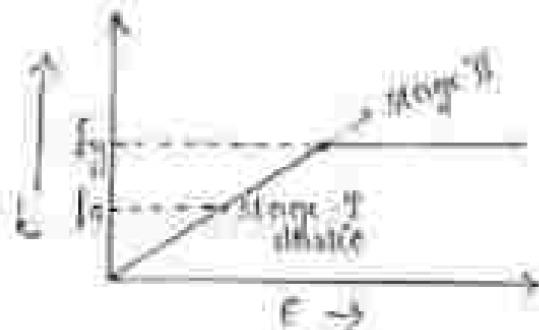
The yield range can be studied more conveniently by increasing the strain scale considerably.



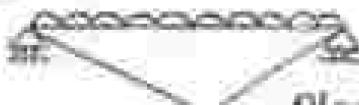
- As the Fig. shows, the plastic range is sufficiently long and it seems reasonable to extend it without limit that is to ignore the effect of strain hardening.
- So the extension of plastic range is supposed to be unlimited at the constant yield stress σ_y or f_y .
- So the idealized elastoplastic stress-strain curve is :-



Bending of Beam:-



Whole section yield = yield stress from extreme fibers to neutral axis (Fully plastic stage). When more rotation will occur.



Plastic hinge

Introduction:-

(i) Steel structures is the application of a group of materials to structural systems to sustain their dead & applied loads.

The design of steel structures involves

(i) Functional design

(ii) Structural design

Functional Design:-

The planning of the structure for specific purpose such as ventilation, lighting, suitable size (size)

Structural Design:-

It consists of proportioning various elements of the building so the most economical manner so that the loads acting on it can transferred safely to the ground without using extra material.

The members are usually subjected to axial forces depend on location on the consideration of all loads. Axial force or other tends to compress members subjected to tensile force by tension members.

(a) - Tension

Members subjected to compression by one compression member.

(a) - Column central

Members subjected to bending by several members.

(a) - Beam

- Advantages of Steel:**
 - Strength of steel structures per unit area.
 - High resistance to high strength pressure and compression.
 - The high strength of steel results in smaller sections to create the same resistance to resist strong loads & unit areas.
 - Possess a high ductility property especially at low temperatures.
 - Possess smooth surface & sufficient long length adhesion.
 - Structured steel are tough & also have both strength and ductility. This means that when a load is applied to the structure the member will not break the structure.
 - Due to weight steel structures can be conveniently handled & transported.
 - Properly designed steel structures have a long life.
 - The properties of steel doesn't change with time.
 - This makes steel the most suitable material for structures.
 - Possesses a resistance to work easily with other structures.
 - They can be treated at a faster rate.
 - Steel is relatively inexpensive material.

Disadvantages:

- It is susceptible to corrosion therefore they need frequent painting & maintenance.**
 - The steel structures need labour to maintain.
 - It has a high cost of construction as compared to other materials.
 - Maintenance cost is also high.

- Poor fire protection as at 1000°F it softens steel.
- 100% of strength remains. The strength decreases with increase in temperature.

- Electricity may be required during insulation.

Notes:-

- Composite construction of steel & concrete can also be used however the majority of present day structures consists of R.C.C or steel.

Uses of Steel structures:-

- The use of steel in building material has been increased now-a-days.

- Bridges
- Dams
- Highrise buildings
- Industrial buildings
- Transmitter towers

Structural Steel :-

- It is an alloy of iron & carbon. In a standard structural steel carbon contains by volume 0.2 to 0.35%.

- Structural steel may be classified by the Bureau of Indian Standards (B.S.I) based on different yield strengths.

Physical Properties:-

Physical properties largely depends on chemical composition, rolling methods, heat treatment & strain hardening.

O Modulus of Elasticity (E):-

Q) Other modulus :- (C or E)

$$G = 2.760 \times 10^3 \text{ N/mm}^2$$

Q) Poisson's Ratio :- (μ)

i) \rightarrow Large range - 0.5

Plastic range - 0.5

ii) ≈ 0.33

St/1

Q) Tensile strength (σ) :-

$$\sigma_a = 1000 \text{ kg/cm}^2$$

Q) Yield point of steel (f) :-

$$f = 1500 \text{ kg/cm}^2$$

Chemical composition :-

• Chemical composition of iron & the allied elements carbon, sulphur, manganese & silicon etc. thus carbon has maximum influence on the physical & mechanical properties of steel.

• True carbon steels containing upto 2% carbon are called carbon steel & those having more than 2% carbon are called cast steel.

• With increase in carbon the tensile strength increases but the ductility falls & thereby the steel is less malleable.

• If the carbon content is reduced the steel will be softer & more ductile but also less malleable by adding manganese, nickel, vanadium (Cr), the tensile strength tends to decrease while retaining the required ductility.

Rolled Steel Sections-

In the design process one of the main object is selection of the appropriate cross section for the desired number of functions. So it is more convenient when a standard cross sections shape is available which can be used in design as well as manufacture and economic fabrication.

The standard sections of structural shape of steel is formed by hot rolling and cold rolling.

Structural steel can be rolled into various shapes & sizes.

Sectons having larger module of section in proportion to their cross-section are preferred.

$\Rightarrow T/4$

Steel sections are named according to their cross-section shape.

Round and square shape are rapidly available in market due to its large number higher values regular steel section.

Some commonly used other section are :-

Other Steel (I. M. S.)

\Rightarrow Square Beam (T.S.P)

\Rightarrow I-beam weight Beam (T.S.P)

\Rightarrow Medium weight Beam (T.S.P)

\Rightarrow Light weight Beam (T.S.P)

\Rightarrow Wide flange Beam (T.S.P)

Special Section:-

\Rightarrow Tension Members (T.M.)

Light channel (T51C)

Middle weight channel (T51C)

c) Angle Section :-

Equal angle (T5A)

Unequal angle (T5A)

Bulb angle (T5B)

d) T-Section :-

Double T-section (T5T)

Light T-section (T5LT)

Short Flange T-section (T5ST)

Heavy Flange T-section (T5HT)

Normal T-section (T5NT)

e) Rolled Steel Bar:-

Square bar (T59A)

Round bar (T59G)

f) Rolled Steel Tubular Sections:-

Light weight tubular section

Medium weight tubular section

Heavy weight tubular section

g) Rolled steel plate (TS)

h) Round steel strips (T55T)

i) Rolled steel flats (T5F)

T-section :-

It is designated by its overall depth & weight
T5T 150 @ 57.7 N/mm²



Uses of columns:-

TSR, TSMR, TSWB & TSJB are used as beam sections & TSJB is used as column.

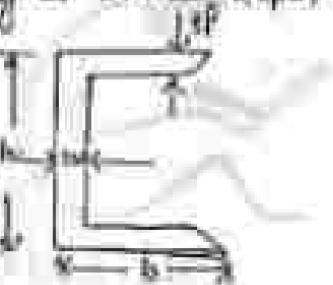
Channel Sections:-

It is designated by its overall depth & weight
ex:- 153E 50 @ 56.9 N/m.

Uses:-

Are used as beams & columns.

For heavy columns built up channels are used.



Angle Sections:-

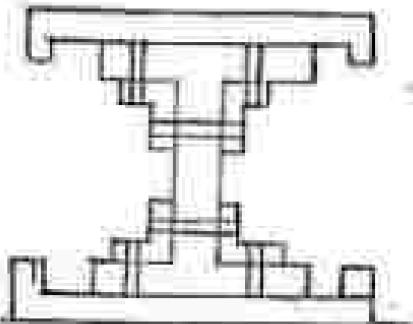
It is designated by its length & thickness of leg.
Ex:- 15A 50x50x6mm

These sections are special sections and are used in ship buildings.



Uses:-

Compression members, tension members & component part of built up members.



In-Section:-

It is designated by overall depth & weight.

Ex:- T61 @ 31.2 kN/m

T5LT 50 @ 31.2 kN/m

U-Section:-

Compression member, Tension member,
frames of doors & windows.



Rolled Steel Bars:-

A round bar is designated by its diameter while as a square bar is designated by its side.

Ex:- 15x12

15x12

Rolled Steel Tubular Sections:-

It is designated by its outside diameter and wall weight.

Ex:- Circular hollow sections

Square

Rectangular hollow sections



Classification according to real stresses:-

Rigid Steel plates:-

It is designated by length, width & thickness.

Ex:- 1000 x 1000 x 10



Reinforced Steel Plates:-

These are designated by width & thickness.

Ex:- 1000 x 100 x 10 mm



This designation is same for strips.

Plates:-

T.S.R & T.S.MS are the only T-acting bars used in India.

All standard T-Beams & Channels have a slope of the inside face of flange of 10/3%.

Joints:-

The parts that act on a structure are called joints.

For the safe design of a structure it is essential to have a knowledge of various materials or members or combination of loads acting on it.

Design Philosophy:-

Design of steel structures consists of designing individual members & their connections.

so that they can safely and reasonably resist and transfer the applied load to the ground.

Learn

The design process begins with selection of material and choosing its safety.

This is where different approaches to design come in the picture:-

The design of structures and elements are based on standardised "Initial yielding".

Assumption of full yielding

- Tensile strength
- Bending buckling
- "Yield" displacement permitted
- Stress Concentration
- Fatigue
- Creep

The design philosophy can have a varied basis, depending upon their selection :-

- 1) Working stress Method
- 2)Ultimate load Method
- 3) Limit State Method

Working Stress Method:-

It is the most "stringent" that the material can withstand while being stretched or pulled before failure or fracture.

Yield Strength:-

It is the stress at which the stress-strain curve becomes non-linear, indicating deviation from Hooke's law of $E = \frac{\sigma}{\epsilon}$. From the linear elastic zone on the stress-strain curve becomes non-linear.

Working stress Method:-

It is the usually adopted of design.

Owing to this method, the members are designed

Permissible Stress according to test

The permissible stress is defined as the ratio of yield stress according to test to safety factor.

$$\boxed{\text{Permissible Stress} = \frac{\text{Yield Stress}}{\text{Safety Factor}}}$$

Limitations:-

According to the method - factors lead to failure of structure and increasing loads which are two

$$\boxed{\text{Failure load} = \text{Working load} \times FOS}$$

Reliability is a more famous in material science than load with yield approach at a fibre

• In structures just formation of plastic hinge is not the collapse criteria - here it can exist till some more hinge formed resulting into collapse mechanism

- It gives unidirectional action
- It deals only with ductile behaviour of material
- The strength of the section at the working load is obtained from the yield stresses of the section

Maintainability

The method is simple & reasonably reliable

Ultimate Load Method:-

From the strain - stress curve it is observed that higher loads than elastic method can be applied on the structure.

This is due to the fact that a majority part of the curve lies beyond the yielding limit.

- This strength is called designed stress; & based upon this strength plastic design is made.
- This method is based on failure conditions rather than working load condition.
- The strength of the section is estimated from ultimate strength of the section.
- In plastic design method, the working loads are multiplied by a load factor and the collapse load and members are designed on the basis of collapse strength.
- Given, the section load should be less than the collapse load by a factor of safety & the members designed should be safe.

Advantages:-

Distribution of external forces is anticipated & considered.

Disadvantages:-

It doesn't guarantee serviceability performance like deflection, instability, creep, cracks & fatigue etc.

So to take care of design requirements from serviceability & sustainability criteria, limit state method is developed.

Limit State Method:

• It is similar to plastic design which considers most critical limit state of strength & serviceability.

• The acceptable limit for the safety & serviceability requirements before failure occurs is called limit state.

• The section design should satisfy serviceability requirements such as limitations of deflection & it should not collapse under accidental loads.

Limit state of Strength :-

* For ensuring the strength capability of structures the loads are multiplied by relevant load factor (γ_f) given in the codes. Code 467-11

* The modified loads are called Factor loads required for the consideration due to the variability of magnitude of load and the loads.

* The design strength of members or its capacity is influenced by dividing ultimate strength supported safely factor (γ_u). The materials used in the code test - trials.

Limit state of deformability :-

It is the limit state beyond which the structural member such as deflection, vibration, temperature damage due to impact, torsion, fire resistance are no longer valid.

Load factor (γ_f) is one factor for all types of their deformability requirements.

Code for Loads :-

IS 457 : part 3 (dead load)

IS 878 : part 2 & 4 (live load)

IS 875 : part 3 (wind load)

IS 875 : part 4 (seismic load)

Mechanical Properties of Steel :-

1) Fluidity

2) Plasticity

3) Durability

4) Strength

5) Hardness

6) Elasticity

Malleability :-

Property of material can be rolled or bent into thin sheet without breaking.

It can be bended back, stretched & twisted upto a high stress before failure.

- (i) Yield deformation
- (ii) Yield stress
- (iii) Ultimate stress
- (iv) Percentage elongation

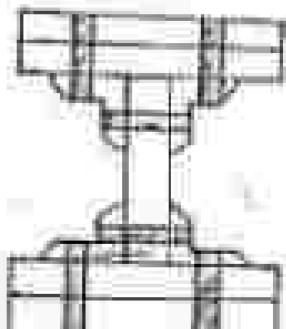
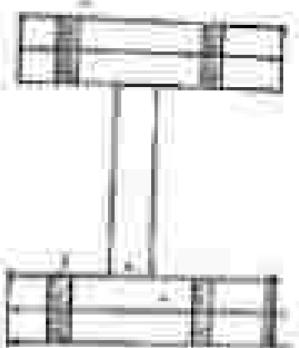
STRUCTURAL STEEL CONNECTIONS

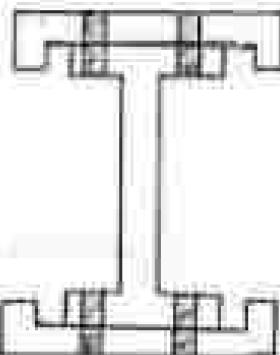
Various elements of a steel structure like tension compression & flexural members are connected by joints or connectors.

The need for designing connections are :-

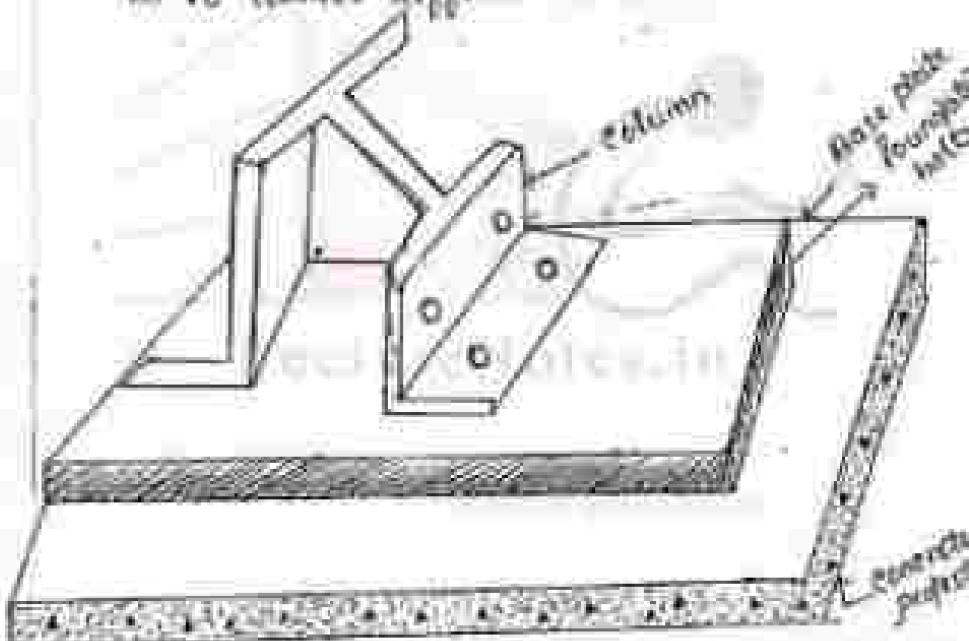
→ Different sections to form the required built up or composite section of a member.

→ It connects plates + angles, channels & section.





(ii) To connect different members of the frame.



(iv) connections of two lengths of a member to make up a required length

If the necessary connections are inadequate the result will be a poor structure inspite of the most efficiently designed members

Therefore, design of connections is very important because the failure of joints is sudden and violent.

The various types of connections used in steel structures are :-

- 1) Rivet connection
- 2) Bolt connection
- 3) Weld connection

Riveted connections :-

When members of a structure are connected using rivets the joints so far is known as riveted joint and the process of joining is known as riveting.

Rivet is made up of a round shank steel bar or body called shank and a head at one end.

Types :-

Since the analysis & design of a riveted connection are same as that for orthotropic bolt, the design of rivets may be done similar to bolts.

Classification based on shape of rivet head:-

i) Snap head rivet



ii) Pan head rivet



iii) Flat counter-sunk



iv) Round counter-sunk



Classification based on method of placing of rivets:-

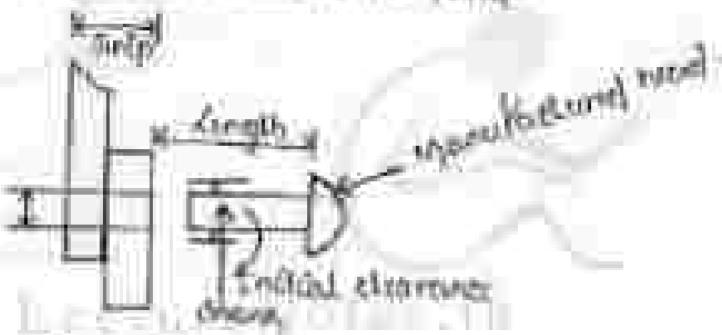
Rivets driven shop rivets:- The rivets which are driven

by hydraulically in the step rivet, control riveting.
Hand driven shop rivet :- The rivets which are driven by hand in the shop.

Cold rivet :- The rivets which are driven at the place of work.

Hot driven rivet :- When the work are heated to red hot before driving they are known as hot driven rivet.

Cold driven rivet :- These are driven at room temperature and high pressure is required to form the head which is not visible to test in the field.



ϕ = nominal diameter.

d = grip diameter.

Grip length \geq the diameter of rivet

Disadvantages:-

- It is associated with high cost of rivet position.
- It needs heating the rivet to red hot.
- Preparation of connections required skill workers.
- Labour cost is high.

• Bolted Connection :-

A bolt may be defined as a metal pin with a head at one end & a shank threaded at the other end to receive a nut or washer as fig.



Bolts are used for joining together pieces of metals by inserting them (bolts) through hole in the metal & tightening the nut at the threaded ends.

Types :-

- Bolts
 - Unfinished bolts / Grade bolt
 - Finished bolts / Turned bolt
 - High strength friction grip (H.S.F.G)

Unfinished bolt :-

It is made from mild steel and with square or hexagonal head.

Nominal diameters are 12, 16, 20, 25, 30 & 36 mm.

This were designated as M12, M16, M20, M25, M30 & M36.

IS 1061 gives specification for such bolts.
Yield strength is equal to 240 N/mm²
Ultimate Strength is 400N/mm².

Uses:-

Light structures, suspending connections.

Friction bolts-

It is made from mild steel bar form for
hexagonal bar & twisted to a circular shape.

Actual diameter is larger than the nominal
diameter i.e. 2 mm to 1.5 mm.

Bolt hole is always bigger than the nomi-
nal diameter of bolt.

To avoid reverse the application.

Clips-

Special joint like connecting machine parts and
used in dynamic loading.

Type:-

It is made from high strength steel and it has
a polished finish.

The bolts are tensioned by using calibrated
wrenches and nuts are provided by clamping device.

In this bolts bearing load is reduced by
frictional force but the number is more & what

To avoid reverse the application.

Nominal diameters are 10, 12, 14, 16 & 18.

Clips-

Forces which subjected to significant bending
classification of bolt joint as per practice:-

(i) Panel type

(ii) Guided type

Guided type:-

It is a transverse type which is made
by bearing.

With the 2 fingers

- ① Unfinished
- ② Finished

Friction Type:-

The force is transferred by friction between member & bolt.

$$F_c = \mu N F_s$$

Classification of bolted connection:-

• On the basis of classification of resultant force transfer:-

→ Concentric connection

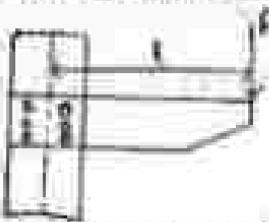
→ Eccentric connection

→ Moment resisting

Example:-

→ axially loaded, tension & compression members.

Bolted connection & slab connection.



Slab-column connections in framed structures

• On the basis of classification of type of joint

① Glue joint:-

When the load is transferred to shear.

e.g.: Lap joint & butt connection.

② Friction connection:-

In this, load is transferred through the friction. E.g.: Hanger connection.

Scaphoid, Shear & Tension Connection :-

C.S. Scaphoid of Bolts :-

* On the basis of bolt mechanism :-

+ Reaction type :-

• Bolt force against the hole to transfer the force.

• Bolt force is transferred through adhesion.

• Mixing of bolts.

+ Friction type :-

• When the load is transferred by friction but the pitch due to loosening of the bolts.

Note:-

The ratio of net tensile area of threads to area of plain shank area of bolt is 0.75 (according to IS 807 part 2).

Tensile Stress

As per IS 807 net tensile area is the area of rest of the threads.

It is called stress area or proof area.

Table 19 of IS code 800:2007 gives stresses for bolt holes.

The bolts of property class 4.6 & 8.8 are given available.

Most common is black bolt of class 4.6.

The no. before decimal indicates yield strength of the material. whereas tensile strength is the no. after decimal indicates the ratio of yield stress to ultimate stress expressed as %.

σ_u = Yield stress

$$UTS = 400 \text{ N/mm}^2$$

$$\sigma_c = UTS/1.15$$

$$\sigma_s = \sigma_c \times 1.05 = 424 \text{ N/mm}^2$$

1.05

$$\sigma_b = Yield \ stress$$

$$UTS = 400 \text{ N/mm}^2$$

$$\sigma_d = UTS$$

$$\sigma_e = UTS \times 1.05 = 424 \text{ N/mm}^2$$

1.05

Specification for spacing:-

* 'P' should not be less than 2.5d
where, d = nominal dia. of bolt.

* 'P' is not more than 3d or 300mm; whichever is less

P \geq 1.5t
300mm ? tension

P \geq 1.5t
300mm ? compression

where,

t = thickness of inner plate

In staggered pitch, pitch may increased by 50% of
value specified value above - the provided gauge distance
is less than 115%

Tensile of butt joint

1) max pitch is to be restricted 4.5d

2) for a ~~max~~ distance of 1.5 times the
width of plate from the bedding surface -

Within gauge length 'l' should not be more
than 100mm or 300mm whichever is less

P \geq 100mm or 300mm

(iv) Min^{mm} edge distance to min. 1.7 x dia. of hole
 instead of other & hand plate cut edges.
 with 1.9 times \times dia. of hole instead of max
 storage test.

(v) tensile $\geq 13tE$ where $E = \sqrt{390/t}$,

to neglect where t = thickness of thinner
 corrugated plate.

Plates in a joint which with a bending of bolts may
 under tension fail due to three causes:-

- 1) Shearing of edge;
- 2) Crushing of plate;
- 3) Rupture of plate.



The shearing & crushing failures are provided for
 the min^{mm} edge & end distances as per IS 800 recommendations
 are provided.

Rupture Failure:

Tensile strength of plate of joint against rupt-
 ure : $Tan = \frac{c_1 A_{eff}}{l_{eff}}$ (p-12, 6-34)

where, A_{eff} = net effective area of the plate at critical
 section.

For ultimate shear of the plate -

Tensile force at ultimate stress -

$$\Delta P = (b - 2d_h) \sigma_t$$

$$\Delta P = \left[(b - 2d_h + 2, \frac{D}{d_h}) D^2 \right] t$$

where - b = width of plate.

d_h = no. of bolt holes.

t = thickness of base plate.

D = diameter of bolt hole.

Design of Strength of Bolt -

1) Shearing capacity of bolt.

2) Tensile capacity of bolt.

Slipping capacity of bolt -

Designing shearing strength of bolt -

$$V_{ult} = V_{ultb} / f_{ultb} \quad (P = F_{ultb} / A_b)$$

$$V_{ultb} = P_{ultb} (n_{ultb} + 2n_{shb})$$

($P = F_{ultb} / A_b$)

where : F_{ultb} = ultimate shear strength of a bolt.

n_{ultb} = no. of shear planes with threads intersecting the shear plane.

n_{shb} = no. of shear planes without threads intersecting the shear plane.

A_{shb} = threaded plane shear area of the bolt.

A_{ultb} = net shear area of the bolt at threads.

Reduction Factor for Shearing Capacity of Bolt -

1) If the load is too long -

2) If the pitch length is large -

3) If the packing plate is stiff -

Bearing Capacity of Bolt:-

$\text{N}_p = \text{N}_{pb}$

F_b

$F_b = \text{FOS of bolt material}$

$$\text{N}_{pb} = 0.5 \times \text{F}_b \times \frac{\text{P}}{\text{d}_b}$$

where, N_{pb} is a factor depends on $\frac{\text{P}}{\text{d}_b}$, $\frac{\text{P}}{\text{D}}$ - 0.35, $\frac{\text{P}}{\text{D}}$ - 0.25.

where,

$\text{F}_b = \text{Ultimate Strength}$

$\text{P} = \text{Pitch diameter}$

$\text{d}_b = \text{dia. of bolt hole}$ (T-13, T-14)

$\text{F}_u = \text{Ultimate Strength of bolt}$

$\text{F}_u = \text{Ultimate Strength of plate}$ (T-17, T-18)

Specification Of bolt :-

Nominal size of bolt - 12, 14, 16, 20, 22, 24, 30-36

Dia. of hole - 13.10, 16, 20, 24, 26, 32, 38

Color code of washer - 30, 31, 34, 35, 36

Graduation Of Bolt :-

Grade	$\text{R}_y (\text{N/mm}^2)$	$\text{F}_u (\text{N/mm}^2)$
U-T	460	460
U-S	330	430
S-S	300	500
S-H	160	500

Efficiency Of Joint (η) :-

It is the ratio of strength of joint / design strength of joint to the ultimate strength of plate

It is always expressed as.

$$\eta = \frac{\text{Strength of joint}}{\text{Strength of plate}} \times 100$$

Terminology :-

① Pitch :-

It is the centre to centre spacing of two bolts in a row measured along the direction of bolt.

② Gauge height :-

It is the distance between the lower edge of a bolt head and is measured at right angles to the direction of bolt.

③ Sag distance :-

It is the distance of centre of bolt hole from the top edge of plate.

④ End distance (E.D) :-

It is the distance of the nearest bolt hole from the end of the plate.

⑤ Stepover distance :-

It is the centre to centre distance of stepped holes on adjoining members.



stepover pitch.



- (2)
- ① When T' should not be more than $\frac{1}{2}T$, i.e., $T' \leq \frac{1}{2}T$
 - When T' should not be more than $\frac{1}{2}T$, otherwise it can be done by successive approximation.
 - if $T' > \frac{1}{2}T$ then $T' = \frac{1}{2}T$.
 - ② In case of half joints, maximum height will be $\frac{1}{2}T$.
 - ③ In case of square joints if height is to be more than $\frac{1}{2}T$ then $T' < T$ otherwise $T' = T$.
 - ④ Maximum height of column
 - a) If $T' > T$ then all in case of alternating column joints will be equal.
 - b) If $T' < T$ then all in case of corner joints, corner joints will have single height. - ⑤ Maximum height of column
 - a) If $T' > T$ (\downarrow - diagonal of corner plate).
 - b) If $T' < T$ (\downarrow - diagonal of corner plate) - ⑥ Calculate the strength of a corner stiffener of square tube for the following position. The main plates to be joined are shown below.
 - i) Top joint.
 - ii) Single corner with joint, one corner plate being 10 mm thick.
 - iii) Double corner or II type of corner plate being 2 mm thick. - Ans: Assuming the factor of safety = 1.5
 - $J_{11} = 100 \text{ kg/mm}^2$ ($T=20, T'=10$)
 - $J_{22} = 100$ ($T=20, T'=10$)
 - Strength of wall
 - $J_{33} = 100 \text{ kg/mm}^2$ ($T=20, T'=10$)
 - $d = 20 \text{ mm}$
 - $A_s = 20 \times 2 = 20 \text{ mm}^2$ ($T=20, T'=10$)

Max. shear stress at mid-span

(2)

⑥ max. shear of wall ($\sigma_{xx}, \sigma_{yy} > 0$)

$$V_{wall} = \frac{V_{total}}{2L}$$

$$V_{total} = \frac{\pi D^2}{48} (v_1 A_{xx} + v_2 A_{yy})$$

$$v_2 = 0$$

$$v_1 = 1$$

$$A_{xx} = \pi D^2 / 4 = \pi D^2 / 4 \times 10^{-6} \text{ m}^2$$

$$A_{yy} = \frac{D^2}{4} \times 4^2$$

$$V_{total} = \frac{\pi D^2}{48} (1 \times 210 + 0)$$

$$\approx 20.59 \text{ kN}$$

$$V_{wall} = \frac{V_{total}}{2L} = \frac{20.59}{1.6 \times 10^3} \approx 12.86 \text{ kN}$$

⑦ max. shear of wall ($\sigma_{xx}, \sigma_{yy} < 0$)

$$V_{wall} = \frac{V_{total}}{2L}$$

$$V_{total} = \sigma_{xx} \times \pi D^2 / 4 + \sigma_{yy} \times \frac{\pi D^2}{4}$$

v_2 is negative sign of v_1 because

$$(i) \frac{v_2}{2L} = \frac{\sigma_{yy}}{2L \times 2} = -0.5$$

$$(ii) \frac{v_2}{2L} = \frac{\sigma_{yy}}{2L \times 2} = -0.5 = -0.5$$

$$(iii) \frac{v_2}{2L} = 0.5$$

$$(iv) \frac{v_2}{2L} = 0.5$$

$$\therefore v_1 = 1$$

$$\therefore V_{total} = 0.5$$

$$V_{total} = 0.5 \times 210 \times \pi D^2 / 4 + 0$$

$$= 105 \text{ kN}$$

$$V_{wall} = \frac{V_{total}}{2L} = \frac{105}{1.6 \times 10^3} = 78.125 \text{ kN}$$

minimum edge distance

$$d = 1.5 \text{ m}$$

$$= 1.5 + 2.2 + 2.2 = 6 \text{ m}$$

maximum gap

$$g = 0.1 \times d$$

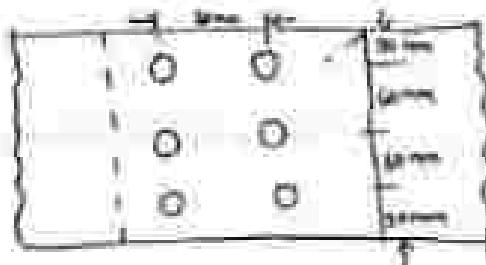
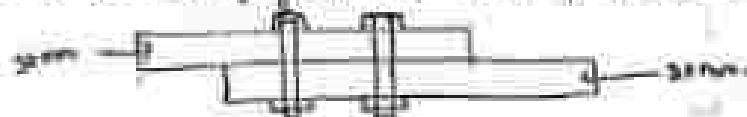
$$= 0.1 \times 6$$

$$= 0.6 \text{ m}$$

Max. shear of the wall is equal to min. val. of the above

Design strength of the wall is equal to 1.1-SDS area (400).

E Find the efficiency of leg joint shown in the figure
from the ratio of reaction to the live load and wind.



S: Find the total load capacity of the beam.

$$P = 20 \text{ kN}$$

$$Q = 22 \text{ kN}$$

$$R_{\text{wind}} = 400 \text{ N/mm}^2$$

$$J_{\text{min}} = 1.85$$

For Fe 400 Grade:

$$f_y = 400 \text{ N/mm}^2$$

$$E = 200 \text{ GPa}$$

$$Z_{\text{min}} = 1.1$$

$$J_{\text{tot}} = 1.85$$

Design strength of steel plate (P-37, G-24)

$$T_{\text{des}} = \frac{0.9 \times 400 \times 100}{J_{\text{tot}}} = \frac{0.9 \times 400 \times 100 \times 10^3}{1.85} = 632.000 \text{ KN}$$

$$t_m = \left[\frac{100 - 100}{100} + \frac{\frac{P_e^2}{R_e^2}}{4 \cdot \mu} \right] z_l$$

$$\approx (100 - 3.72 + 0) \times 30$$

$$\approx 2520 \text{ mm}^2$$

Design Strength of Wall

C bending strength of wall ($P = 300, I = 2.2$)

$$V_{wall} = \frac{V_{app}}{f_{ck}} = \frac{271.72 \text{ kN}}{300} = 0.9057 \text{ kN/mm}$$

$$\begin{aligned} V_{app} &= \frac{f_{ck}}{I} (n_{app} + n_{app}) \\ &= \frac{100}{2.2} (2 \times 0.975 \times \frac{2}{3} \times 20^2) \\ &= 222.44 \text{ kN} \end{aligned} \quad \left(\text{from } f_{ck} = 100 \text{ N/mm}^2 \right)$$

D Shear strength of wall ($P = 300, I = 2.2$)

$$V_{app} = \frac{V_{app}}{f_{ck}} = \frac{100 \cdot 300}{300} = 100 \text{ kN/mm}$$

$$V_{app} = 2.5 \times 100 \times 0.975 \times \frac{2}{3} = 2.5 \times 0.975 \times 20 \times 20 \times 0.975 = 122.25 \text{ kN}$$

n_{app} is load of V_{app}

$$\sqrt{\frac{f_{ck}}{f_{ck}}} = \sqrt{\frac{100}{100}} = 1.00 \text{ (from } f_{ck} = 100 \text{ N/mm}^2 \text{)}$$

$$\therefore \frac{f_{ck}}{f_{ck}} = 0.975 \text{ (approx)}$$

$$\therefore V_{app} = 100 \text{ kN/mm}$$

bending Shear strength of concrete = 222.44 kN/mm

strength of joint is equal to sum of 3 values ($= 211.00 \text{ kN}$)

equation of joint :- (300 N)

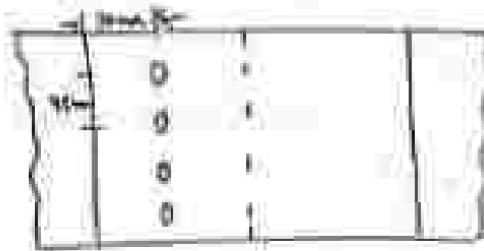
$$\frac{\text{strength of joint}}{\text{strength of wall block}} = \frac{211.00}{222.44} = 0.947 / (Ans)$$

strength of wall block ($P = 300, I = 2.2$)

$$T_{app} = \frac{V_{app}}{f_{ck}} = \frac{100 \times 300 \times 20^2}{300} = 200 \text{ kN}$$

For a single twisted cable when we just do not do connect two plates when we do not want to increase the ratio of grade 46 and 33 plates to be 6 to one which calculate the strength and efficiency of the joint, if 4 rolls are provided in the both then we get the result as shown in fig.

(2)



Let's assume the two grades of steel.

$$f_y = 400 \text{ N/mm}^2$$

$$f_{y2} = 330$$

For 46 grade of steel.

$$f_{y2} = 400 \text{ N/mm}^2$$

$$f_{y2} = 1.8 f_y$$

$$f_{y2} = 1.8 \times 400$$

$$= 720 \text{ N/mm}^2$$

$$f_y = 400 \text{ N/mm}^2$$

$$f_y = 30 \text{ mm}$$

$$d = 20 \text{ mm}$$

Thickness of tension plate = 3 mm.

Strength of tension plate = $\frac{\pi d^3}{48} f_y$ (from, 11-14)

Strength of tension plate = $\frac{\pi d^3}{48} f_y$ (from, 11-14)

$$T_u = \frac{\pi d^3}{48} f_y$$

$$= \frac{0.9 \pi (3 - 3)^3 \times 400}{48}$$

$$= \frac{0.9 (36 - 27) \times 3.14 \times 400}{48}$$

$$= 63.26 \text{ kN.}$$

Strength of wall

(1)

D. Tension strength of wall

$$\text{Strength} = \frac{\text{tensile stress}}{\text{stress ratio}} = \frac{0.3 \times 10^6}{1.5} = 0.2 \times 10^6 \text{ N/mm}^2$$

Wall = $\frac{1}{2} \pi r^2$ (in ft & in ft)

$$= \frac{\pi r^2}{4} (12 \times 10^{-3} \times 10^6 \times 10^{-6})^2 = 12.56 \text{ kips}$$

B. Shearing strength of wall

$$\text{Strength} = \frac{\text{shear stress}}{\text{stress ratio}} = 0.1 \times 10^6 \text{ N/mm}^2$$

$$\text{Strength} = 0.001 \times 10^6 \times 10^6 = 0.1 \times 10^6 \times 10^6 = 0.1 \times 10^6 \text{ N/mm}^2$$

C. Bond shear strength

$$\text{Strength} = \frac{\text{shear force}}{\text{area}} = \frac{F_s}{A_s}$$

$$(0.1) \frac{F_s}{A_s} = 0.25 \times \frac{F_s}{25 \times 10^{-3}} = 25 \times 0.1 \times 10^6 = 25 \text{ kips}$$

$$\therefore F_s = 25 \text{ kips}$$

∴ Strength of slab is equal to sum of above 3 values i.e. 0.1 + 25 = 25.1 kips

Efficiency of the slab =

$$\eta = \frac{\text{Strength of slab}}{\text{Strength of slab plate}}$$

$$\frac{25.1 \times 10^6}{0.001 \times 10^6 \times 10^6} = 25.1 \text{ times}$$

Strength of slab plate



$$\text{Strength} = \frac{\text{Shear force}}{\text{Area}} = \frac{0.1 \times 10^6 \text{ N/mm}}{100 \times 10^{-3}} = \frac{10 \times 10^6 \text{ N/mm}}{100 \times 10^{-3}} = 100 \times 10^6 \text{ N/mm}^2$$

Now we can calculate the shear force diagram. The shear force starts at zero at the left end and increases linearly to a maximum value of 200 N/mm . Then it remains constant until the point where the shear force becomes zero again.

For the left quarter of the beam:

$$\Delta y = 100 \text{ mm}/\text{mm}^2$$

$$T_{xy} = 100 \text{ N/mm}$$

$$T_{yy} = 100 \text{ N/mm}$$

$$T_{xx} = 200 \text{ N/mm}$$

$$\text{Shear force} = \frac{\text{Total moment}}{\text{Length of one beam}}$$

Example of solution:

Calculating moment of inertia:

$$I_{xx} = \frac{T_{xx}}{T_{yy}} = \frac{200 \text{ N/mm}}{100 \text{ N/mm}} =$$

$$I_{yy} = \frac{T_{yy}}{(T_{xx} + T_{yy})} =$$

$$= \frac{100}{200} (100 + 200) = 50 \text{ N/mm}^2$$

$$= 50 \text{ N/mm}^2$$

So the bending stress and deflection are not linear. Therefore when we are calculating bending moment diagram, it is different from the linear theory. Considering bending moment diagram, it is non-linear with respect to deflection.

Let's assume the both to be linear:

Deflection of beam DD

Bending moment of beam DD

$$\text{Deflection} = 50 \text{ N/mm}^2$$

$$= 50 \text{ N/mm}^2$$

$$= 50 \text{ N/mm}^2$$

Deflection	Bending moment
0	0
0	0
0	0

(4)

$$\Rightarrow p = \frac{q_0 + 2\gamma \cdot \eta_{mf}}{\eta - q_0 + 2\eta_{fr}} = \frac{q_0 + 2\gamma \cdot 1.35}{\eta - q_0 + 2 \cdot 10} \\ \approx 21 - 32 \text{ mm}$$

Wet pitch = 2.5 cm

$$= 2.5 \times 10^{-2} \text{ m}$$

$$= 25 \text{ mm}$$

will provide the same pitch.

$$\text{existing edge distance (c)} = \frac{q_0 - 2\gamma \cdot \eta_{fr}}{\eta} \\ = 48 \text{ mm}$$

check against bearing strength

K_b is read off the following

$$\Rightarrow \frac{c}{3d_b} = \frac{40}{25 \times 2} = 0.64$$

$$\Rightarrow \frac{c}{2d_b} = 0.32 = \frac{65}{25 \times 2} = 0.25 < 0.334$$

$$\frac{K_b \cdot d_b}{10} = 0.925$$

$$\Rightarrow K_b = 0.464$$

$$\text{Value} = \frac{25 \times K_b \cdot 20 \times 2 \times 0.5}{J_b}$$

$$\therefore \frac{25 \times 0.464 \times 20 \times 0.5}{J_b} = 410 \\ \therefore 79.004 \text{ kN} > 410 \text{ kN}$$

So design is OK.

Q. Two plots of size 2 ha each are to be ploughed by a tractor, working at a rate of 1 ha per hour. The tractor driver has just got the following data:-
 Tractor's speed limit = 10 km/hr.
 Speed of tractor = 8 km/hr
 Speed of tractor = 6 km/hr
 (Actual speed of tractor = 8 km/hr).
 (Cost per hectare = Rs. 100/-).

Ques:- Total area

For a 10% margin of field

$$\text{Area} = \frac{100}{100+10}$$

$$= 100 \text{ m}^2$$

$$= 10 \times 10 \text{ m}^2$$

$$= 100 \text{ m}^2$$

For the two plots of land

$$\text{Area} = 2 \times 100 \text{ m}^2$$

$$= 200 \text{ m}^2$$

$$\text{Tractor's actual speed} = \frac{\text{Actual speed}}{100+10} \text{ km/hr.}$$

$$= \frac{8}{100+10} \times 100 = \frac{8}{110} \times 100 = \frac{800}{110} = 7.27 \text{ km/hr.}$$

Time taken



Time required to plough one plot of land = $\frac{\text{Area}}{\text{Speed}}$
 = $\frac{100}{8} = 12.5 \text{ hours}$.
 Time required to plough both plots = $2 \times 12.5 = 25 \text{ hours}$.
 Cost = $25 \times 100 = \text{Rs. } 2500$.

$$\begin{aligned} \text{Time required} &= (\text{Actual speed}) \times \text{Time} \\ &= (8 - 100) \times 25 \\ &= 1.2 \times 25 \\ &= 30 \text{ hours} \end{aligned}$$

Ans :-

Estimated time = 30 hours.

Reason: Reasoning example of the field is given above and by ploughing the entire surface area of the field, we get the work done in the field.

Chung, Pham, & Lai

$$V_{\text{air}} = \frac{\text{Volume of vessel}}{\text{Time}} = \frac{100 - 61.6}{1.33} = 29.7 \text{ ml/sec}$$

$$V_{\text{eff},n} = \left\{ \frac{1}{\sqrt{2}} \left(n_1 | 0000 \rangle + n_2 | 1111 \rangle \right) \right\} \cdot P_{\text{eff},n}$$

$$= \frac{1}{12} \left(17.372 \times 2.5^4 + 100.000 \times \frac{2}{3} \times 2.5^2 \right) = 9.$$

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also the value = $\frac{1}{\sqrt{2}}$

150, 4-06
92-75

1	2	3	4	5	6	7
o	o	o	-	o	o	o
o	o	o	-	o	o	o
o	o	o	-	o	o	o

Get μ to the pink region.

Strength of the links from P_1 to P_2

$$T_{\text{kin}} = \frac{0.94 m v}{T_{\text{ext}}}$$

$$\sim \frac{e^{-\eta} e^{(p-q_0)t + i k W}}{\sqrt{2\pi t}} \quad (t = 10^{-m})$$

$$= \pi_2 \cdot q c_2(p-1,1) \neq 0$$

strength of two bodies from which nothing

• 1991. 9월 • 82. 4호

W. E. Gandy by G. F. W.

$$2.952(1.12) = 1.85 \cdot 9.6$$

(ii) $\rho = 92 \text{ mm}$

$$\text{Min pitch} = 2\pi d \\ \approx 50 \text{ mm}$$

Max pitch = diameter of bolt $\approx 30 \text{ mm}$

Let's provide a pitch $\approx 30 \text{ mm}$

Check against bearing strength

$$e = 1.25d \Rightarrow 1.1750 \approx 33 \text{ mm}$$

R_s is least of the following.

$$(i) \frac{\sigma}{E} = 0.81^2 = 0.91 \quad (ii) \sigma^{0.22} \\ (iii) e = 1$$

$$(iv) \frac{L}{2d} = 0.15$$

$$\therefore R_s = 0.9$$

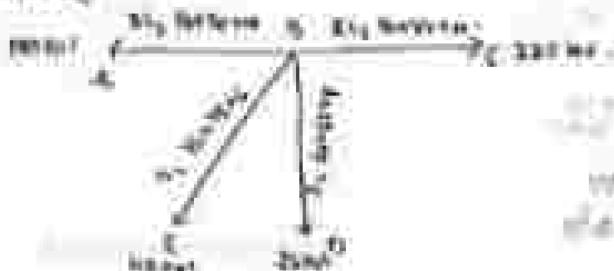
$$V_{app} = \frac{2 \times \pi \times R_s \times d \times \rho \times v}{J_{app}}$$

$$\approx \frac{2 \times \pi \times 0.9 \times 20 \times 10 \times 70}{1.95}$$

$$\approx 92.98 \text{ kN}$$

Mass of ball = 0.05 kg
Initial velocity = 10 m/s
Time taken by ball to reach the ground = 2 s

Ans: 100 J



(1)

Ques: Calculate:

i) Kinetic energy of ball.

$$K.E. = \frac{1}{2} m v^2 / \text{Joule}$$

$$m = 0.05 \text{ kg}$$

$$v = 10 \text{ m/s}$$

$$K.E. = \frac{1}{2} \times 0.05 \times 100 = 2.5 \text{ Joule}$$

ii) Potential energy of ball at

$$\text{At } 10 \text{ m height}$$

$$P.E. = mgh / \text{Joule}$$

Energy of ball in air = 100 J

$$\text{Energy} = \frac{\text{Initial Energy}}{\text{Time}} = \frac{100}{2} = 50 \text{ Joule}$$

Energy of ball in air = 50 J

Energy of ball in air = 50 J

$$K.E. = \frac{1}{2} m v^2 / \text{Joule}$$

Energy of ball in air = 50 J
(Assume $m = 0.05 \text{ kg}$)

$$50 = \frac{1}{2} \times 0.05 \times v^2$$

$$50 = \frac{1}{2} \times 0.05 \times \frac{v^2}{0.05} \Rightarrow 50 = 0.05 v^2 \Rightarrow v^2 = 1000$$

$$\therefore \frac{v^2}{0.05} = 10000$$

$$\therefore K.E. = 5000 \text{ Joule}$$

(i) $t = 6\text{ m}$.

$$\text{Voltage} = \frac{100}{6} = 16.67\text{ V}$$

(iv)

(ii) $t = 3\text{ m}$.

$$\text{Voltage} = \frac{100}{3} = 33.33\text{ V}$$

(iii) $t = 12\text{ m}$.

$$\text{Voltage} = \frac{100}{12} = 8.33\text{ V}$$

(v) $t = 15\text{ m}$, V ?

$$\text{Voltage} = \frac{100}{15} = 6.67\text{ V}$$

Question No. 7 :-

Find the load \sim (in ohms).

The machine is composed of double angle section ICA 30 x 100 mm and is connected on the opposite sides of a 15 m long南北 pole. The pole will be 15 cm thick and will bear against the iron sheet (rest of 15 cm and 20 cm \rightarrow air gap).

Hence strength of the coil will be $1000 \times 15 \times 10^6$ ampere turns.

i.e. 15000.

$$\text{No. of coils required} = \frac{100}{15000} \Rightarrow 0.0067 \approx 7 \text{ coils.}$$

Question No. 8 :-

Find the load \sim (in ohms).

The machine is composed of double angle section ICA 30 x 100 mm and is connected on one opposite side of a 15 m long南北 pole. The pole will bear against the iron sheet (rest of 15 cm and 20 cm \rightarrow air gap).

Hence strength of the coil will be $1000 \times 15 \times 10^6$ ampere turns.

i.e. 15000.

$$\text{No. of coils required} = \frac{100}{15000} = 0.0067 \approx 7 \text{ coils.}$$

Explain how to do it.

- The solution is single angle solution as it is a right angle and it
is resulting from a common side given that the two sides is single
angle being equal to each other (total of given and known)
∴ Strength of bolt will be sum of 30 Nm and 20 Nm i.e. 50 Nm
∴ M.M. of bolt required = $\frac{50}{\pi \times 25^2} = 0.0128 \approx 0.013$

Explain how to do it.

Probability based on the data.

- The solution is single angle solution as it is a right angle and it is
resulting from a common side given that the two sides is single
angle being equal to each other (total of given and known).
∴ Strength of bolt will be sum of 30 Nm and 20 Nm i.e. 50 Nm
M.M. of bolt = $\frac{50}{\pi \times 25^2} = 0.0128 \approx 0.013$.

Explain the basic concept of bolt used in connecting two plates
to beam in fig. Then
↳ Very difficult to disengage at later time
↳ - difficult to fit

→ Solution of bolt of grade 8.8 will be discussed below in
different types of fastener used.

Explain how to do it.

D	D ₁	t	O	=	D ₂
10	8	1	8	=	10
12	10	1	10	=	12



Explain how to do it.

For 2.5 grade of soil

$$\text{Area} = 100 \text{ m}^2$$

Thickness of 1.6 mm

$$q_u \approx 15 \text{ kN}$$

For 100% safety ($P = 75\%$, $M = 50\%$)

$$V_{ult} = \frac{V_{dry}}{Y_{ult}}$$

$$V_{dry} = D_s \cdot Y_{dry} \cdot K_d \cdot V_{sat}$$

$$D_s = \text{coefficient of saturation} = 0.8$$

$$K_d = (\text{soil density under field}) \times 2$$

$$K_d = 1.0 \text{ (constant value)}$$

$$C_s = \text{angle of internal friction}$$

$$\approx 30^\circ \approx 2.7 \text{ rad} \approx 0.55 \text{ rad/m}$$

$$V_{sat} = 100 \text{ kN/mm}^2$$

$$V_{ult} = 1.10 \text{ (if safety factor is designed at seismic peak)} \\ + 1.95 \text{ (if safety factor is designed at wind load)}$$

$$\text{Design } V_{ult} = 2.3 \times 0.75 \times 2.7 \times 0.55 \\ = 53.45 \text{ kN}$$

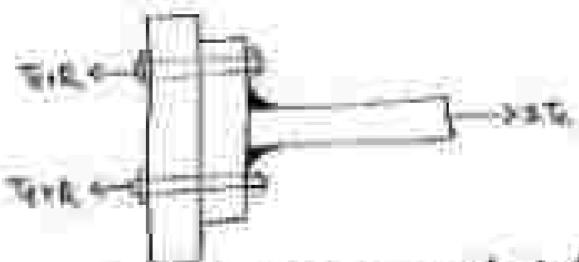
(i) If the resistance designed at seismic load,
but no design shear capacity

$$V_{ult} = \frac{V_{ult}}{\gamma_{ult}} \cdot \frac{\gamma_{ult}}{\gamma_{des}} = 53.45 \text{ kN}$$

(ii) If the resistance designed at seismic load,

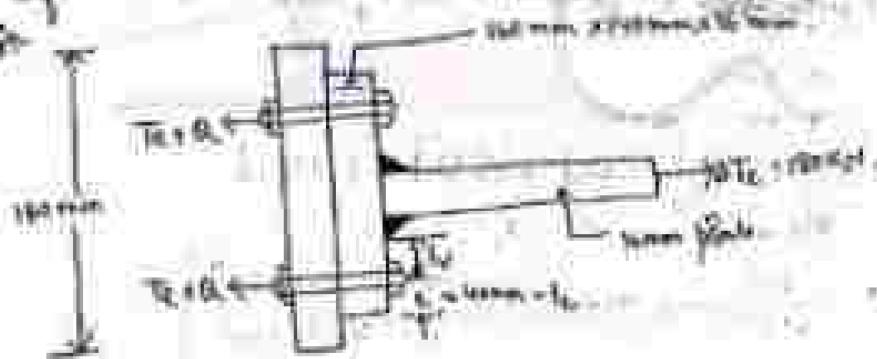
$$V_{ult} = \frac{V_{ult}}{\gamma_{des}} = \frac{53.45}{0.75} = 69.93 \text{ kN}$$

(7)



If at elevation $T.L.$ there is cavitation the head of z h.
There will be no additional head as is provided to reduce the
head required to prevent cavitation occurring in the pipe.

b) The jet shown in the figure has to carry a head of z h.
The total head and end of the stream stage one is 16 m . The water
surface and end of the stream stage two is 14 m . Then condition one stage is
water surface and



∴ the elevation of the free end is 14 m

$$\therefore l_2 = \frac{160}{2} - \frac{14}{2} = 40 - 7 = 33 \text{ m}$$

Pumping head

$$Q = \frac{l_2}{2h} \left[T_L - \frac{3\pi f_s h^2}{274 l_2 l_1^2} \right] \quad (P-32, 9-4-7)$$

$$L = \sqrt{\frac{P_{n_0}}{f_0}} = \sqrt{\frac{27.51}{0.17}} = 21.83 \text{ mm}$$

P = 1 (Because 45% are always positioned left) (16)

$$T_{n_0} = 15^\circ\text{C}$$

$$f_0 = 0.702 \text{ Hz}$$

$$\sim 0.0025 \text{ g} = 25 \text{ N/mm}^2$$

$$f_{n_0} = 250 \text{ N/mm}^2$$

$$c = 16 \text{ mm}$$

Calculated by the formula given

As = load of the building

$$(2 \times 1.141) \sqrt{\frac{0.10}{0.17}} = 26.53 \text{ mm}$$

$$\therefore c = 16 \text{ mm}$$

$$\therefore l_n = 26.53 \text{ mm}$$

As = effective width of flange per pair of bolts = 100 mm

$$2T_{n_0} = 100 \text{ N/mm}$$

$$\therefore T_{n_0} = 50 \text{ N/mm} = 50 \times 10^3 \text{ N/mm}$$

$$Q_n = \frac{l_n}{2l_n} \left[l_n - \frac{p_{n_0} b c t}{24 E I_y} \right]$$

$$= \frac{54}{97.56 \cdot 2} \left[97.56^2 - \frac{13 \times 10^3 \times 100 \times 16}{27 \times 24 \times 13 / 24} \right]$$

$$\approx 36.47 \text{ kN}$$

Design shear strength of bolt

$$T_{n_0} = \frac{0.972 A_{n_0} v}{f_{n_0}}$$

$$= \frac{0.972 \times 1.5 \times 27 \times 4 \times 230}{150} = 146.53 \text{ kN}$$

Tensile load on the bolt = $T_{n_0} + R_n = 160.53 \text{ kN}$

$T_{n_0} < T_{n_0} + R_n = 160.53 \text{ kN} < T_{n_0}$

Welded connection is made by joining two members in such a way that there is no relative movement between them.

- The members are joined by single plates and the weld is made by means of electric arc or by submerged arc welding along with flux. which adds strength to the joint.

Advantages of welded joint :-

- ① Welded joints offer the opportunity to obtain a more efficient structural design than the riveted joints in the same structural members.

- ② The cost of fabrication being higher (approximately 3 times) than riveted joints due to the overhead cost of welding equipment and labour.

- ③ Welded joints are better for repeated use of structures.

Disadvantages of welded joint :-

- ④ The weld connecting the plates may be dangerous, relatively weak than rivets.

- ⑤ The joints connecting up the weld are rigid and hence different stresses reflected.

- ⑥ Due to stress and its related loads are increased. effects of fatigue stresses, stress concentrations and shape of the weld are reflected.

Types of welded joint

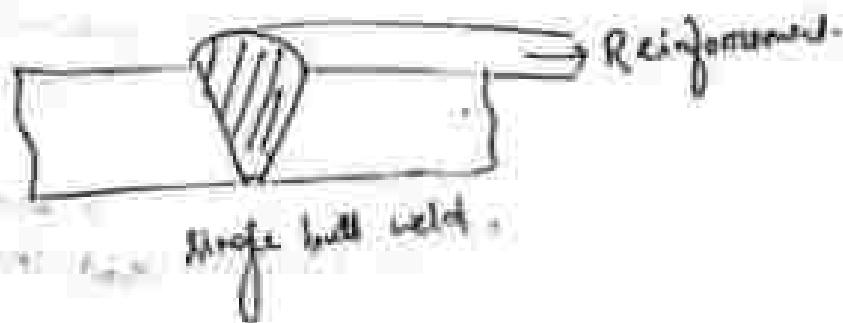
- ① Butt weld on plates with
- ② Fillet weld. ③ Plug weld.
- ④ Edge weld.

① Bell weld :-

Bell weld is also known as groove weld depending upon shape of the groove made for welding. And weld configuration

is single bell weld & double bell weld.

②



Specification of setting



Front - Reel + Te
+ Te

Size of reel :- (P-30, M-T-3)

- The size of the reel will be sufficient to accommodate the reel wire.
- The distance between the reel and reel + setting will be equal to the reel size.

Max. size of reel :-

"The max" size of a reel will be obtained by subtracting 1.5 mm from the diameter of the thread number to be joined.

Minimum size of reel :-

The size of the reel number.

Size (mm)	Reel and setting (mm)	Reel size (mm)
0	10	3
10	20	5
20	32	8
30	40	10

Effective front clearance (P-30, M-T-3)

It is the distance between from the end of the front setting to the free end of reel.

- The effective front clearance should not be less than 3 mm and it must not exceed 0.75 or 1.5.

effective front clearance = X X S

Design strength

σ_d is the design of the joint with respect to which the strength of the joint must satisfy with each of the design requirement.

Q



(a)

Design strength of gross width :-

The design strength of gross width is taken as uniform
The thickness of joint is taken as uniform

$T_{gross} = \frac{t}{T_{gross}}$

This is effective length of joint for gross width.

It is effective length of joint for gross width.

Joint is treated as fully welded

- for the top welding

- for the side welding

Design strength :- (gross width)

In case of complete penetration of the gross width, design calculation can take into account of the gross width as the joint is treated as not required as the width because all the joint is required to the strength of the member concerned.

In case of incomplete penetration of the joint width, the effective joint length is required and the required effective length is determined by which the strength equal to the strength of the member concerned.

Design strength strength of joint width :-

$J_{width} = \frac{t}{T_{width}}$

for a given design of joint $\sigma_d = \frac{f_y}{f_y}$

The design strength of a joint width is based on the strength of the joint in which $P_{gross} = P_{width} = \frac{f_y}{f_y}$

minimum effective length of the wire for the sake of the safety
with a condition of $\sigma_{max} = \sigma_{allow}$ which are given.

- (i) The case arising when an intermediate girder will stand at
carrying the full superimposed load for some time. Then it is
to take the value of σ_{max} .
- (ii) The negligible intermediate girder will stand for a longer period
less than the value of σ_{max} as the stresses will
gradually decrease.

gives us the following form :-

$$\frac{\sigma}{\sigma_{max}} = \frac{L}{L + l_0}$$

$\sigma_{max} \rightarrow$ calculated working stress - $L \rightarrow$ effective overall
length of beam \rightarrow σ_{allow}
 $\sigma \rightarrow$ Stress stress \rightarrow effective length of beam
 $\sigma_{max} \rightarrow$ permissible stress

Failure of member :-

- (i) Buckling :-
When the axial load is increasing to such an extent that
the member, the section through the width is increased to
such an extent that it is causing the failure to occur
in axial and the process occurring known as collapse



The force of the muscle is increased by increasing the length of the muscle along the elongated form the rest of the muscle.



(ii)

Contraction of muscle

Muscle fibre is contracted

- (i) When muscle is contracted
- (ii) When muscles are more rigid or compacted, so contraction of muscle can contract more, less pliable, stretching under the effect along with the muscle during long contraction and make the joint more flexible
- (iii) Due to force of your muscle fibres, a tendon becomes shorter & shorter, when you are contracting opposite your tendon is lengthened
- (iv) Muscles can be contracted for helping in case of walking, jumping, running
- (v) The power of walking is greater in comparison to running
- (vi) The power of walking is short, whereas in case of running it is longer.
- (vii) In case of walking, the leg bones are required for the weight bearing
- (viii) In walking the leg bones are required for the weight bearing in one straight direction & in case of running the leg bones are required for the weight bearing in two directions
- (ix) In other places, both are alike i.e. in both cases, the skeletal muscles are required for the contraction and relaxation of muscle fibres.

- Q) The offspring of which bird is more likely to inherit a
cancer gene? (a) jungle babbler (b) house sparrow
- (b) Cancer is an inherited trait. Due to the high cancer
incidence among humans, there will be an increased risk in their
offspring.
- (d) The frequency of a birth defect is more in the case of male
birds as compared to female and female birds
- (e) The capture of certain birds is difficult and expensive. Hence
breeding such birds can be helped easily by applying the
method of cloning.
- (f) A more trained person is required to care a wild bird
compared to a wild/ domestic bird.

Q) Two pairs of wings and three wings have to be joined by a
glue and as shown in fig. The pair is supposed to be joined
pairs from a group. One has to join either the upper pair of
wings with the middle pair or the middle pair with the lower pair. Then the
wings of the pair of

(i) Single V wings will be produced

(ii) Double V

Wings are likely to be destroyed

(g) Data given:

Area same for two groups of birds.

by = 20 cm²

length of wing = L = 100 cm.

For bird A: Y = 100 cm

V-jointed weld



Show incomplete penetration of weld takes place.

$$\therefore \text{Cross thickness} - t_c = 0.757 \times 4 \\ = 0.757 \times 14 \\ = 10.57 \text{ mm.}$$



$$\text{Strength of weld } T_{w2} = L_w \times t_c \times \frac{f_y}{J_w}$$

$$= 138.4 \times 9.89 \times \frac{340}{1.95} \\ = 346 \text{ kN} < 430 \text{ kN.}$$

So it is unsafe.

Diff. double V-jointed weld

Show complete penetration of weld takes place.

$$\therefore t_c = 14 \text{ mm.}$$

Strength of weld [LectureNotes.htm](#)

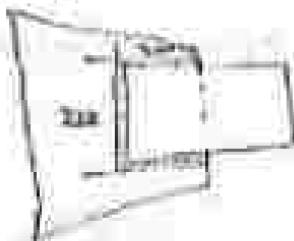
$$P_{w2} = L_w \times t_c \times \frac{f_y}{J_{max}}$$

$$= 138.4 \times 14 \times \frac{340}{1.95} = 490 \text{ kN} > 430 \text{ kN.}$$

∴ So it is safe.

Ex 4. A member in a frame subjected to $200 \text{ mm} \times 75 \text{ mm}$ is shown. It is welded to a $15 \text{ mm} \times 15 \text{ mm}$ gusset plate by a fillet weld. If the member is 300 mm and has width $b_1 = 15 \text{ mm}$, determine the design strength of the joint if the welding is done as shown in fig which is the same as strength of the joint if welding is done all along the top edge.

(2)



Sol: For the joint grade of weld,

$$d_w = 300 \text{ mm}$$

$$d_y = 150 \text{ mm}$$

$$\text{Joint factor } K_{w,y} = 0.87$$

The effective length of weld $L_e = 2(300 + 150) - 2(15)$
 $\approx 820 \text{ mm}$

Effective throat thickness $t_e = 15 \text{ mm}$
 $\approx 0.75 \text{ in}$
 $\approx 19.1 \text{ mm}$.

Design Strength of weld, $P_{w,d} = 1.62 \times 24 + \frac{10}{100}$

$$= 383.44 + \frac{100}{100} = 383.54 \text{ kN}$$

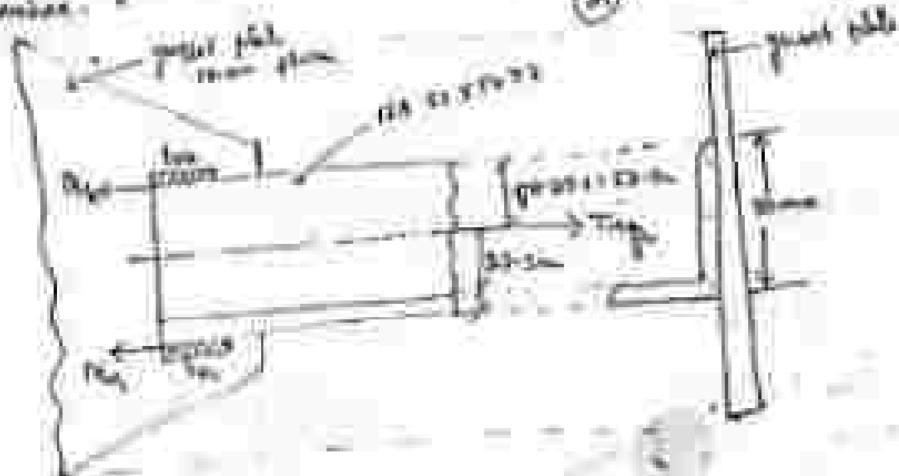
When the welding is done all along,

$$L_{e,1} = 2(300 + 150) = 900 \text{ mm}$$

$$P_{w,d,1} = 1.62 \times 24 + \frac{10}{100} \times \frac{900}{300} = 413.54 \text{ kN}$$

Strength in the direction of the joint = $413.54 - 383.54 = 30 \text{ kN}$

4. The section (I-beam section) weighing 15.8 kg/m = 0.775 m³ (0.75 grade steel) is welded to a corner plate, which plate also - design width to transmit load equal to the design strength of member.



Data given

for 0.75 grade steel,

$$f_y = 40 \text{ kN/mm}^2$$

$$f_u = 60 \text{ kN/mm}^2$$

Factor safety factor against yielding $f_{y,s} = 1.1$.

For 0.75 width : $f_{y,s} = 0.75 \times 1.1 = 0.825$

From Ductile Analysis for the section (D.A.C.)

$$A_g = 1732 \text{ mm}^2, C_g = 375 \text{ mm}^2$$

The design strength of the section governed by yield of gross section.

$$T_{w,d} = \frac{A_g f_y}{f_{y,s}} (1.0) \quad (1.0)$$

$$\frac{1732 \times 40}{0.825} = 2125.27 \text{ N}$$

∴ The width will be designed to transmit a force of 2125.27 N.

(a) $T_{\text{eff}} = \text{Strength of fiber with no tensile force resisted by width of effective length } l_e$

(b) $T_{\text{eff}} = \text{Strength of fiber with no tensile force resisted by width of effective length } l_e$

Taking moment about line of action of T_{eff}

~~Equilibrium condition~~

$$T_{\text{eff}} \times 250 = T_{\text{eff}} \sqrt{(25 - 21.5)}$$

$$\Rightarrow T_{\text{eff}} = 145.75 \text{ N/mm}^2$$

$$T_{\text{eff}} + P_{\text{ext}} = T_{\text{app}}$$

$$\Rightarrow P_{\text{ext}} = T_{\text{app}} - T_{\text{eff}}$$

$$= 75.55 \text{ N/mm}^2$$

(b) Size of fiber with (a) :-

Actual size of one fiber width = 15 mm.

Now = size of one = $\frac{1}{15} \times 15 = 1 \text{ mm}$

Let's provide 6 mm size of fiber width.

effective strand thickness = $t_s = 0.756$
~~mm~~

average thickness of width

$$\text{from } t_{\text{avg}} = \frac{l_e \times t_s \times b}{\sqrt{3 \pi f_{\text{avg}}}}$$

$$\Rightarrow l_e = \frac{t_{\text{avg}} + 6.5 t_{\text{avg}}}{t_s + b}$$

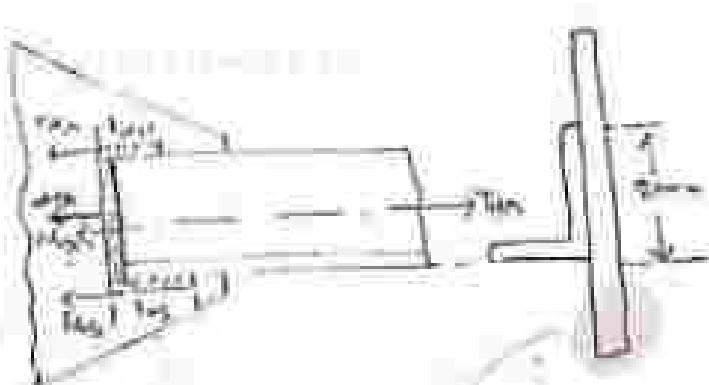
$$= \frac{145.75 \times 75 \times 1.15}{1.15 \times 150} = \frac{1093.125}{225} = 4.86 \text{ mm}$$

$$t_{\text{avg}} = \frac{l_e \times b \times 3}{\sqrt{3 \pi f_{\text{avg}}}}$$

$$\Rightarrow t_{\text{avg}} = 13 \text{ mm}$$

(b)

Cut the soil with 10 cm width of the trench
 (10 cm width of soil) is taken in a square
 hole of size of the soil is the same as the mass per



\therefore the area of square = 100 cm^2

$$d_s = 10 \text{ cm}$$

for side wall, $L_s = 1.5$

$$\text{Total wall height} = L_s + d_s + h_s = 1.5 + 10 + 10 = 21.5 \text{ cm}$$

Tensile strength of soil

$$T_{soil} = \frac{\text{dry wt}}{L_s} \times \frac{\text{dry density}}{100} \quad (L_s = 1.5 \text{ m}, \rho_d = 1.3 \text{ ton/m}^3, P_c = 27.2 \text{ ton})$$

$$= 200.07 \text{ kN}$$

\therefore $T_{soil} = \frac{P_c}{L_s} \times L_s$

$$\frac{200.07 \times 1.5}{1.5} = 300.07 \text{ kN}$$

$$\text{dry wt} = E_s \times V_s$$

$$\text{dry wt} = \frac{V_s \cdot d_s}{L_s} \cdot \rho_d$$

$$S = S \text{ cm}^2$$

$$A_s = 0.176$$

Taking second and bottom fiber.

$$M_{u1} \times 20 + M_{u2} \times 19 = T_{u2} \times 27.3$$

$$\Rightarrow P_{u2} = \frac{292 \times 27.3 - 53 \times 19}{20}$$

$$P_{u1} + P_{u2} + P_{u3} = T_{u1}$$

$$\Rightarrow P_{u3} = 119.92 \text{ kN}$$

$$P_{u3} = \frac{L_u \times 27.3 \times 19}{\sqrt{3} f_{ck} A_u}$$

$$P_{u3} = \frac{L_u \times 27.3 \times 19}{(0.7) f_{ck} A_u}$$

$$\begin{aligned}\Rightarrow L_u &= \frac{P_{u3} \times 19 \times 700}{27.3 \times 19} \\&= \frac{46.37 \times 10^3 \times 10^3}{27.3 \times 19} \\&= 19.12 \text{ mm} \\&\approx 20 \text{ mm}\end{aligned}$$

$$\begin{aligned}\Rightarrow L_u &= 19.12 \text{ mm} \\&\approx 18 \text{ mm}\end{aligned}$$

- Q. In the 300 x 300 mm (Ex 4th grade of steel) L-beam
a portion made from of concrete. The shear strength is to
be reduced at one side to a quasi static value of 100 N/mm. Design
a joint weld, if the overlap is limited to 200 mm.

Sol:- For Ex 4th grade of steel -

$$f_y = 355 \text{ N/mm}^2$$

$$f_{ck} = 35 \text{ N/mm}^2, f_{ctk} = 0.7$$

$$f_m = 300 \text{ N/mm}^2, f_{u1} = 324.7 \text{ N/mm} \quad (\text{P-1b})$$

$$\text{Ans. } A_g = 450 \text{ mm}^2, d_1 = 9.6 \text{ mm}$$

$$d_2 = 6.2 \text{ mm}$$

width of front edge = 5 mm (more than plate).

$$\text{Area} = \frac{1}{2} \times 0.7 \times 15 \\ = 5.25 \text{ mm}^2.$$

Width passing over by front edge = 10 mm.

$$L_1 = 0.7 \times 15$$

$$= 10.5 \text{ mm}.$$

Strength of wood per unit length.

$$P_{\text{max}} = \frac{\text{Area} \times f}{\sqrt{3 + f^2}}$$

$$= \frac{10.5 \times 1000}{\sqrt{3 + 1000}} = 552.33 \text{ N/mm.}$$

Length of wire required $L_2 = \frac{F}{P_{\text{max}}}$

$$= \frac{970 \times 10^3}{552.33}$$

$$= 1754.42 \approx 16.50 \text{ m.}$$

Because of the reduction of 200 mm, working length of 1400

that it be provided in wire say

$$= 5 \times 350 + 200 = 1050 \text{ mm} < 16.50 \text{ m.}$$

Now, let us provide this width.

width of front should be less than 3 times the width of wire

greater

$$3 \times 2 = 6.0 \text{ mm} = 24.35 \text{ mm.}$$

Let's we provide this width \rightarrow let the width of wire

be L.

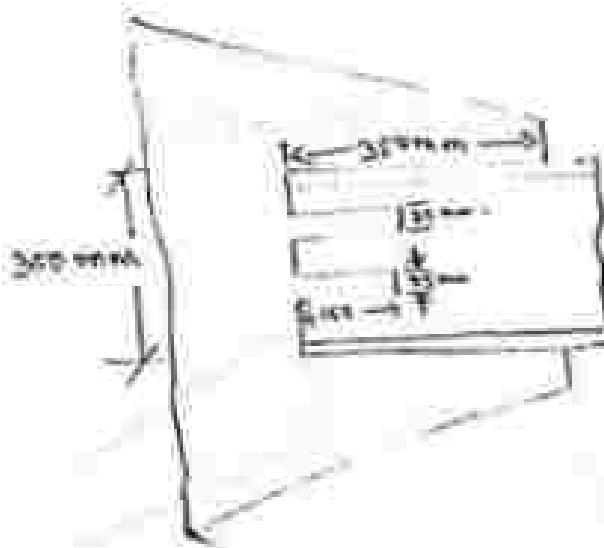
$$1630 = 2 \times 370 + 3x^2 + 4xL$$

(2)

$$\Rightarrow L_1 = 157.5$$

$\therefore 157.5 \text{ mm}$,

Provide 16 mm & 25 mm slot, two in one.



Note:-

Ref to IS 816:1969

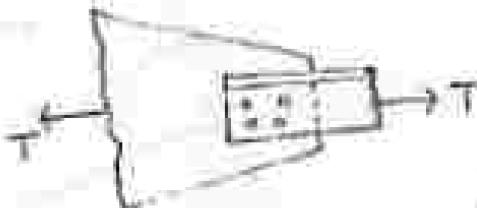
- i. width on side of the slot web should be less than 3 times the thickness of 25 mm which even is greater.
- ii. Slot should be rounded with radius not less than 15 mm.
- iii. Angles on 15 mm which even is ~~more~~ greater.

Design of tension members:-

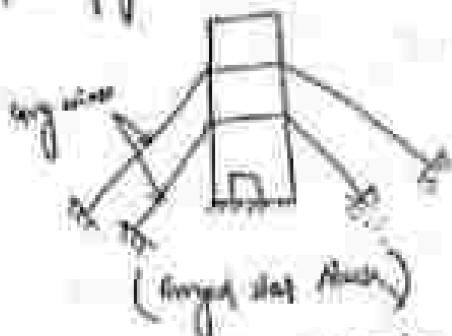
(P-32, section-6)

Tension member:-

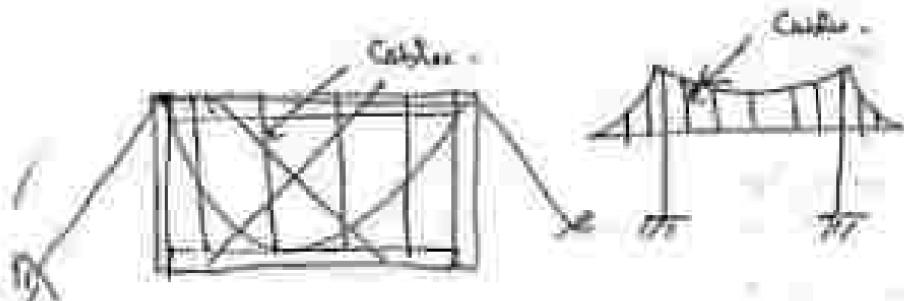
- a) A structural member subjected to the pulling force applied at the ends is called a tension member.
- b) Tension members are also known as tie members.

Types of tension members:-Wire and Cable:-

Wire cables are relatively well for building purposes and we guy wires in steel towers and towers.



Cables used as tie members in suspension bridges are made from individual strands wound together in one large



(b) Bars and Rods :-

These are often used as tension members in building.
Spiral or stay rods to support joints between floors, and
to support girders in industrial buildings.



(c) Plates and Sheet Metal :-

Plates and sheet metal are often used as tension members
in transmission towers, roof bridges etc.

(d) Hollow sections :-

The net sectional area of a tension member

= gross sectional area of the member -

— the sectional area of the maximum
size of holes.

$$A_n = (n - n_{\text{holes}}) A \quad A_n = \left[b - n d_h + \frac{\pi d_h^2}{4} \frac{P_e^2}{E I_{\text{eff}}} \right] A$$

Type of failures :-

① Gross section yielding :-

Concurrent deformation of the members in longitudinal direction may take place before the fracture, making it quite undesirable.

② Net section yielding :-

The collapse of the member when the net cross section of the member reaches the ultimate stress.

③ Buckling failure :-

A segment of beam of material at end of member does not fail due to the failure of high bending stresses of the end and high strength being resulting in multiple unloading regions.

Strenthening Factor (λ) :-

The dimension ratio of a section member to the ratio of the unsupported length L to the radius of gyration r .

$$\lambda = \frac{L}{r}$$

$$r = \sqrt{\frac{I}{A}}$$

Design of Axial member :-

(16)

The design strength of a section member is the lesser of the following.

- Design strength due to yielding of gross section (T_{sg})
- Resilient strength of critical section (T_{rc})
- The buck stress (T_{bs})
- Design strength due to yielding of gross section

$$T_{sg} = \frac{\sigma_y b t}{f_{yec}} \quad (P-32, 6.2)$$

Design of Critical section

(i) For plates

$$T_{rc} = \frac{0.9 f_{yec} t}{2 \alpha_1} \quad (P-32, 6.2.1)$$

(ii) For Flanged ends

$$T_{rc} = \frac{0.9 f_{yec} t}{2 \alpha_1} \quad (P-32, 6.2.2)$$

Δ_c = sum of eccentricities of the flanged sections

(iii) single inflection

$$T_{rc} = \frac{0.9 f_{yec} t}{2 \alpha_1} + \frac{\beta \sigma_y t}{2 \alpha_1} \quad (P-32, 6.2.3)$$

$$\beta = 1.25 - 0.05(\alpha_1)(\frac{t_y}{t_w})(\frac{b}{2c})$$

otherwise

$$T_{dn} = \frac{0.4n^3 u}{J_{ml}}$$

T_{dn} will be head of (iii)

Design strength due to wave shear

$$T_{ds} = \frac{0.4n^3 u}{\sqrt{3} J_{ml}} + \frac{0.9 A_{dm} f_u}{J_{ml}}$$

or $T_{ds} = \frac{0.9 A_{dm} f_u}{\sqrt{3} J_{ml}} + \frac{0.4 n^3 u}{J_{ml}}$

T_{ds} will be head of the above.

Design of tendon reinforcement anticipated to carry load :-

Step-1
Required tendon area is determined by using the formula:

$$A_t = \frac{T f_u}{0.9 f_u} \text{ for tendon.}$$

$T \rightarrow$ factored tendon load.

Step-2 Required tendon area obtained in Step-1 will be increased by 20% - 40% to complete the gross area by

Graph can be drawn by the help of

Tip
By

Slope

From that slope finding the value of T_0 , a initial weight is obtained.

Angle

angle θ can be obtained by using the formula
tan $\theta = \frac{m}{g}$
except of one last

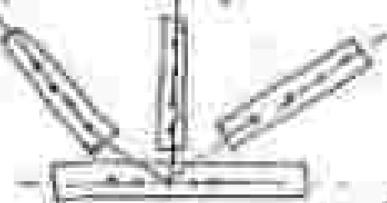
Slope
design stress T_0 of each section is calculated. Then will be minimum of stresses T_0 , T_m & T_{max} .

The design stresses $T_0 > T_m$ & $T_0 < T_{max}$.

Gage plate :-

1 A gage plate is a plate provided to make contact at the time when there was reaction to the joint of two bars under stress.

2 The size and shape of the gage plate are taken from the drawing of the members meeting at joint.

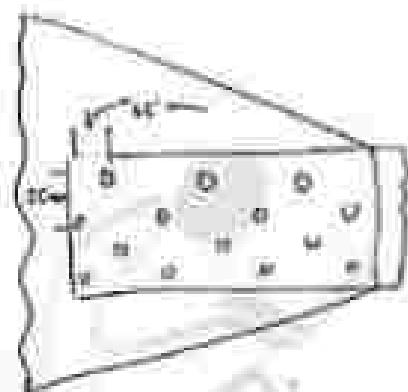
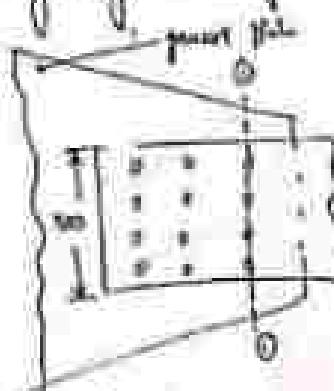


After the action of force when meeting at a joint see

Q. A 300 kg iron of grade 4000 is used to make the tension members in a lattice girder. It is equivalent to a 12 mm dia. gauge wire by 16 mm dia. wire of grade 400. Calculate effective rate area of

constant rolling is done or drawn in zig-

X zig-zag rolling is done "



Sol:- For the two types of rolling,

$$\text{for constant rolling} \quad R = \frac{\pi}{4} \times \frac{D^2 - d^2}{2} \times 10^{-6}$$

$$\text{for zig-zag rolling} \quad R = \frac{\pi}{4} \times \frac{B^2 - b^2}{2} \times 10^{-6}$$

(i) for constant rolling, the initial radius will be $R=1$

$$A_g = (B - r_{\text{min}}) \cdot l$$

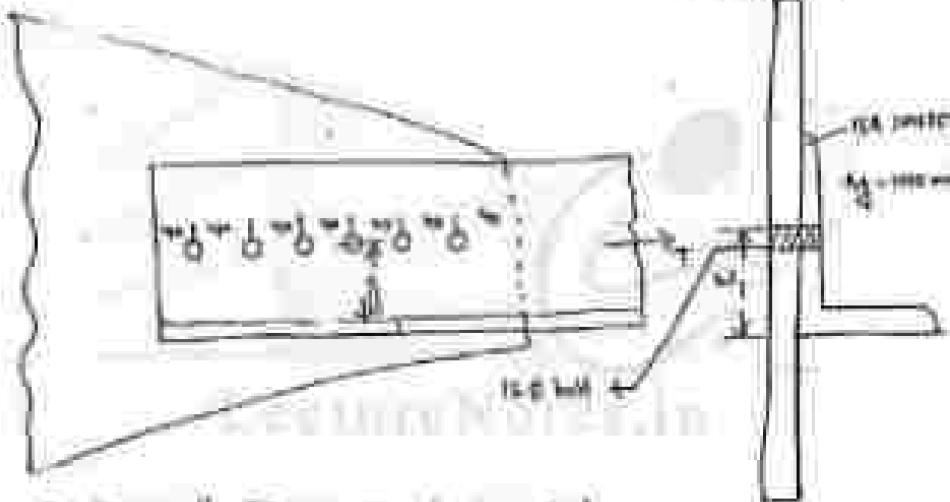
$$\sim (300 - 4 \times 20) \times 10^3$$

$$\sim 1760 \text{ mm}^2$$

E A single unequal angle 100x100x10 is connected to a wall with two equal pins at the ends with a force of 100 N. Determine the axial force in member 100 as shown in fig. Determine the axial force through of the angle assuming that the yield stress value of steel used now 25000 N/mm².

ii) If the pin is connected to the 100 mm by

100 mm



Q. If the TE mm by is connected.

cross of 100 mm by

Find the yield stress of steel.

$$f_y = 250 \text{ N/mm}^2 \quad f_{y0} = 1.1$$

$$f_u = 350 \text{ N/mm}^2 \quad f_{u0} = 1.25$$

$$A = 10 \text{ mm}$$

$$A_s = 10 \text{ mm}$$

(i) Design strength due to yielding of gross area,
(P-32, 6.2)

$$T_{dy} = \frac{A_y f_y}{I_{el}} = \frac{1640 \times 260}{11} \\ = 209,600 \text{ Nm}$$

(ii) Design strength due to collapse of reduced section
(P-33, 6.2)

$$T_{dr} = \frac{\sigma_{red} I_{el}}{I_{el}} + \frac{P_{red} t}{I_{el}} \quad (a)$$

$$I_{el} = \text{net area of connected leg} - \\ \sim (180 - 72 - \frac{L}{2}) \times 6 = 408 \text{ mm}^3$$

$$I_{eq} = \text{gross area of the connecting leg} - \\ \sim (72 - \frac{L}{2}) \times 6 = 432 \text{ mm}^3$$

$$P = 1.9 - 1.076 \left(\frac{L}{2} \right) = \left(\frac{1.9}{10} \right) \times \left(\frac{432}{100} \right) \sim 1.02$$

ω = required leg width $\sim 35 \text{ mm}$.

$$L = 6 \text{ mm} \\ L_{eq} = \text{effective leg width } (I-33, 6.2 c).$$

$$L_{eq} = \text{effective leg width } = 15 + 60 - 6 + 10 = 69 \text{ mm}.$$

$$\omega = 60 \text{ mm } (\text{as 100 mm leg is connected}).$$

$$L_{eq} = 6 \times 60 = 360 \text{ mm}.$$

$$P = \frac{I_0 T_m}{I_0 T_m + I_1 T_m}$$

$$\therefore \frac{I_0 T_m}{I_0 T_m + I_1 T_m} = 0.94$$

$$0.94 \text{ (approx)} < \frac{I_0 T_m}{I_0 T_m + I_1 T_m}$$

$$\therefore P = 0.92$$

$$T_m = \frac{0.9 \times 474 \times 410}{1.25} + \frac{1.02 \times 474 \times 210}{1.1}$$

$$= 210 \text{ Nm}$$

efficiency

$$T_m = \frac{\text{efficiency}}{1.25} (P - \eta_2) = \frac{0.9 \times 906 \times 710}{1.25}$$

$$= 231.73 \text{ Nm}$$

$$\eta = 0.9 \text{ for } 40 \text{ kNm}$$

$$\eta_2 = \text{min value of } \eta_2 \text{ from } 2/12 \text{ to } 11/12$$

$$\sim (\eta_{2c} + \eta_{2o}) = 474 + 470 = 9.46 \text{ m}^{-1}$$

Tension will be least if the following i.e., $240 \times 231.73 \text{ Nm}$

$$\therefore T_m = 231.73 \text{ Nm}$$

∴ Tension due to wind shear $(P-23, 6.4-1)$

$$\text{Wind Shear Strength} = \frac{I_0 T_m \times I_1 T_m}{\sqrt{2} T_m} + \frac{0.9 T_m}{T_m}$$

$$\text{or } T_m = \frac{0.9 T_m}{I_0 T_m + I_1 T_m} + \frac{I_0 T_m \times I_1 T_m}{T_m}$$

$$A_s = (6 \times 40) \times 6 = 1440 \text{ mm}^2$$

$$\text{Area} = \left(C_{740} - C_{718} - \frac{R}{2} \right) \times 6 = 846 \text{ mm}^2$$

$$A_{eff} = (100 - 60) \times 6 = 240 \text{ mm}^2$$

$$A_{eq} = \left(100 - 60 - \frac{R}{2} \right) \times 6 = 126 \text{ mm}^2$$

$$T_{ab} = \frac{1440 \times 840}{\sqrt{3} \times 41} + \frac{0.9 \times 126 \times 410}{1.35} \\ = 243.45 \text{ kN}$$

$$T_{ab} = \frac{0.9 \times 126 \times 410}{\sqrt{3} \times 1.35} + \frac{240 \times 840}{111} \\ = 192.45 \text{ kN}$$

Tension will load of the member ~~243.45 & 192.45~~

$$\therefore T_{de} = 192.45 \text{ kN}$$

Design strength of the member will be least of the following T_{ad} , T_{de} & T_{an}

\therefore design strength of the member is 192.45 kN . Ans.

Given a bridge truss being subjected to a vertical load of 200 kg. The height of the truss is 3 m. The horizontal reaction is considered to a great pitch in the ratio with the ratio of 20 : 1. The safety factor is 2.0.

Design given :-

having the weight of steel.

the S.F. weight of
steel.

$$f_u = 400 \text{ N/mm}^2$$

$$f_y = 235 \text{ N/mm}^2$$

$$\beta_y = 0.55$$

$$\beta_{ymin} = 0.5$$

$$\gamma_0 = 1.1$$

$$\gamma_0 = 1.05$$

$$\gamma_L = 1.25$$

Tensile load $T = 200 \text{ N}.$

Required net area of the angle section

$$A_g = \frac{T \cdot \gamma_L}{0.9 f_y \beta_u}$$

$$= \frac{200 \times 1.25 \times 1.25}{0.9 \times 235} = 1016.26 \text{ mm}^2$$

Required gross area

$$A_g = \frac{T \cdot \gamma_L}{\beta_y}$$

$$= \frac{200 \times 1.25 \times 1.25}{235} = 1324 \text{ mm}^2$$

(Pg. 4)

From the tables,

std's provide 12A 100x15x10 mm as tension member.

$$A_g \text{ provided} = 1324 \text{ mm}^2$$

No. of bolts :-

(i) Shearing strength of bolt in single shear.

$$V_{sp} = \frac{f_b (n_h + n_w)}{T_s}$$

$$= \frac{830 \times 1 + 0.28 \times \frac{\pi}{4} \times 25^2}{13 \times 1.25}$$

$$= 93.94 \text{ kN}$$

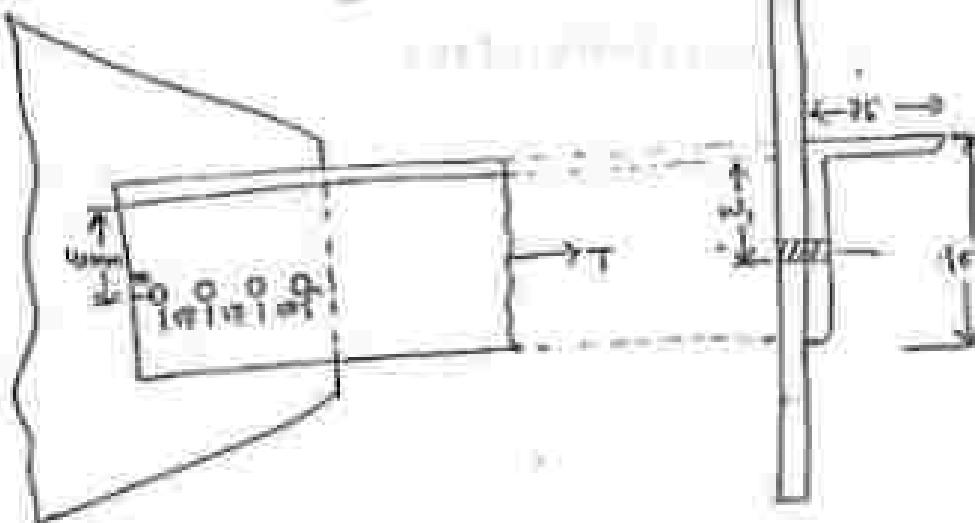
(ii) Bearing strength of bolt

$$V_{pb} = \frac{n_h K_b d t s_f}{T_s}$$

$$= \frac{8.6 \times 1 \times 25 \times 2 \times 100}{1.25} = 131.2 \text{ kN}$$

$$\therefore \text{Strength of bolt} = 93.94 \text{ kN}$$

$$\therefore \text{No. of bolts} = \frac{300}{93.94} = 3.19 \approx 4 \text{ nos.}$$



Ans: provide 4, so cover the ball with 4 layers
 (so we need minimum width 10 mm in one direction). Show in
design sketch for design tensile strength.

(b) Design strength due to yielding (P-32, L-2)

$$T_y = \frac{A_y \cdot f_y}{J_{yy}} = \frac{133.6 \times 250}{111} \approx 303 \text{ kN} > T.$$

So it is safe.

(c) Design strength due to rupture (P-32, L-2)

$$T_u = \frac{W_u A_u f_u}{J_{yy}} + \frac{\text{Plasticity}}{J_{yy}}$$

$A_u = \text{net area of cross-sectional area}$

$$\sim \left(\pi r^2 - \pi r^2 - \frac{\pi}{4} \right) \times Q' \approx 492 \text{ mm}^2.$$

$A_p = \text{gross area of cross-sectional area}$

$$\sim \left(\pi r^2 - \frac{\pi}{4} \right) \times Q' \approx 548 \text{ mm}^2.$$

$$p = 1.44 - 0.0347 \left(\frac{M_u}{M} \right) = \left(\frac{1.44}{M} \right) \times \left(\frac{M_u}{M} \right) = 1.44$$

so, additional longitudinal $\rightarrow 75 \text{ mm}$.

$t = 8 \text{ mm}$.

$b_1 = \text{flange thickness by section}$

$$\sim 1.44 \times 8 - t = 72.448 - 8 = 64.4 \text{ mm}.$$

$w_1 = 40 \text{ mm}$.

$l = 40 \text{ mm}$ and width = 36.17

$$f_t = \frac{f_y I_{max}}{f_y K_{eff}} = \frac{400 \times 10^3}{200 \times 10^3} = 2.0$$

$$1.1 < f_t < \frac{f_y I_{max}}{A_y J_{st}}$$

$$\therefore f_t = 1.87$$

$$T_{dR} = \frac{0.9 \times 290 \times 10^3}{1.25} \rightarrow \frac{107 \times 10^3}{1.25}$$

$$\approx 315 \text{ kN} > 300 \text{ kN}$$

So it is OK.

(iii) Design strength due to static stress: (P.23, 6.4.1)

$$T_{dR} = \frac{f_y I_{max}}{\sqrt{3} J_{st}} + \frac{0.9 f_y A_y J_{st}}{J_{st}} = 315.16 \text{ kN}$$

$$T_{dR_2} = \frac{0.9 f_y A_y J_{st}}{J_{st}} = \frac{0.9 f_y J_{st}}{J_{st}} = 263.16 \text{ kN}$$

$$A_{sq} = (3000 + 40) \times 4 = 1220 \text{ mm}^2$$

$$b_{sq} = \left(3000 + 40 - 3050 - \frac{20}{2} \right) \times 4 = 904 \text{ mm}^2$$

$$b_{sq} = (100 - 40) \times 4 = 240 \text{ mm}^2$$

$$A_{sq} = 0.9 \times 1220 \left(40 - \frac{20}{2} \right) \times 4 = 312 \text{ mm}^2$$

$$\therefore T_{dR} = 263.16 \text{ kN} < 300 \text{ kN}$$

Since T_{sh} is less than T , the number will fall in time trend.

SD

The action can be made safe by increasing the distance of the ball from the wall and by increasing the filter as shown in fig.



$$A_{\text{sh}} = (3 \times 10 + 40) \text{ m}^2 = 104 \text{ m}^2$$

$$A_{\text{in}} = \left(5 \times 10 + 40 - 3 \times 3.3 - \frac{3.3}{2} \right) \text{ m}^2 = 103.4 \text{ m}^2$$

$$A_{\text{sh}} = 2 \times ((10 - 3.3) \times 3.3) \text{ m}^2 = 31.6 \text{ m}^2$$

$$A_{\text{in}} = \left(3.3 - \frac{3.3}{2} \right) \times 3.3 \text{ m}^2 = 4.95 \text{ m}^2$$

$$T_{\text{sh}} = 301.34 \text{ sec}$$

$$T_{\text{sh}} = 301.34 \text{ sec}$$

$$\therefore T_{\text{sh}} = 301.34 > T$$

∴ So it is OK. (Ans).

Compressive Member

(P-34, T-1)

A compression member is a structural member which is straight and subjected to two equal and opposite compressive forces applied at the ends.

For column, joist, girder etc.

Effective length — (P-45, T-2)

Effective length of a compression member is the product of effective length factor 'K' and the actual length 'L'.

mathematically,

$$L_{eff} = K \cdot L$$

The value of 'K' depends upon the supported and relative transverse condition at the end of the member.

for K values (P-45, T-11)

Slenderness ratio: - (P-30, T-10)

Slenderness ratio of a column is defined as the ratio of effective length to concentring radius of gyration of the section.

$$\lambda = \frac{L_{eff}}{r_e} = \frac{K \cdot L}{r_e}$$

\rightarrow critical length of compression member.

\rightarrow effective length.

\rightarrow eccentricity of loading.

Type of Cross-section

- The tendency of a member to buckle is usually measured by the slenderness ratio.

Maximum value of slenderness ratio

$$(r = 50, T = 2)$$

Design of compression members

Step-1 Design limit in compression to be attained.

Step-2

Effective sectional area required is

$$A_e = \frac{P}{f_{ck}}$$

Step-3 Calculate section to give effective area required and calculate Rein.

Step-4 Keeping the end condition and adding the effective connection determine effective length.

Step-5 Find the slenderness ratio and some design stress f_s and find capacity using Eq.

Step-6 Revise the section if calculated f_s differs considerably from the design load.

Design of axially loaded compression Member :-

Assumption

- The axial column is assumed to be axially straight having no initial deflection.
- The modulus of elasticity is assumed to be const. i.e., $E = E_0$.

Design

i. Slenderness ratio is assumed w.r.t. to height of column

Step - 1

For the assumed value of the slenderness ratio i.e. 200-1, the design compressive stress for that value is determined from appropriate curve and working class.

Design Working Stress

The safe working stress required to carry the factored load the actual compressive stress is computed.

$$\sigma_{\text{w}} = \frac{\sigma}{\text{Working Class}}$$

Step - 2

A section that provides the intended required stress is selected from class table.

Step - 3

The effective length of the column is calculated on the basis of eccentricity.

For the estimated value of λ , the design factor, this
can be calculated from Table 4 of page 10.4.1.1.

Example

For a single angle stiffener connecting the stage stamp
to stiffening by using formula

$$\lambda_1 = 0.625 \text{ kip} \quad (\text{P-34, T-1.2})$$

and the design coefficients above using the formula and write

$$\lambda_1 = \frac{\phi/\lambda}{\phi + [\phi^2 - \lambda]^{\frac{1}{2}}} \quad (\text{P-34, T-1.2-1})$$

$$\phi = 0.4 \left(1 + \alpha(\lambda - 0.2) + \lambda^2 \right)$$

$$\lambda = \sqrt{\frac{f_y \gamma \left(\frac{w}{r} \right)^2}{\sigma^2 E}}$$

$\lambda \rightarrow$ Table 4 of P-34.

However, a single angle also twisted through one of the
legs is subjected to flanging torsional bending. The equivalent
flanging ratio is given as

$$\lambda_{eq} = \sqrt{k_1 + k_2 \lambda_{st}^2 + k_3 \lambda_{sp}^2} \quad (\text{P-48, T-1.2-1.1})$$

$(k_1, k_2, k_3 \rightarrow \text{T-12}) \quad t=48"$

$$\lambda_{st} = \frac{1}{\sqrt{\frac{E}{G}}}, \quad \text{and} \quad \lambda_{sp} = \frac{(k_1 + k_2)/2t}{\sqrt{\frac{E}{G}}} \quad (\text{P-48, T-1.2-1.1})$$

Ques-7

The design strength of the column is controlled by formula $P_u = \alpha \times f_y A$.

- E Calculate the design compressive load for a short S235 @ 410.2 N/mm² yield. The column is reinforced in axial and transverse with the bars. Reinforcement area of grade Fe 410.

Ans: Fe 410 grade of steel

$$f_y = 410 \text{ N/mm}^2$$

$$f_y = 235 \text{ N/mm}^2$$

$$\lambda_1 = 1.25$$

From chart table ISMB 300 @ 410.2 N/mm². (P = 14)

$$A_s = 300 \text{ mm}^2 \quad t_3 = 11.6 \text{ mm} \quad t_0 = 56 \text{ mm}$$

$$h_s = 250 \text{ mm} \quad t_{12} = 144.5 \text{ mm}$$

$$z = 121 \text{ mm} \quad t_{12} = 173.2 \text{ mm}$$

$$\frac{L}{b_1} = \frac{300}{300} = 1.0 > 1.2 \quad (I = 47, T = 10) .$$

$$t_3 = 11.6 \text{ mm} < 40 \text{ mm}$$

Working stress with Z=2 and working stress with Z=6.

(54)

Design compressive stress from table 9(a) and
working about Z-Z axis

$$\lambda_z = \frac{KL}{\pi y} = \frac{0.65 \times 3.5 \times 10^3}{145.5}$$

$$\approx 15.52$$

From table 9(b) (P-40)

$$\lambda \quad \text{def} \quad (\text{for } \lambda_y = 9.0)$$

$$10 \rightarrow 297$$

$$20 \rightarrow 226$$

~~$$\text{Design } f_{cd} \text{ (15.52)} = 297 - \frac{297 - 226}{20 - 10} \times (15.52 - 10)$$~~

$$= 226.448 \text{ N/mm}^2$$

$$\therefore P_d = A_e f_{cd}$$

$$= 922 \times 226.448$$

$$= 2082 \text{ kN}$$

Design compressive stress from table 9(b) and working
about Y-Y axis

$$\lambda_y = \frac{KL}{\pi y} = \frac{0.65 \times 3.5 \times 10^3}{52.2} = 43.5 \text{ mm.}$$

From Annexure 9(b) - (P-42)

$$\frac{\lambda}{\lambda_0} \rightarrow \frac{\text{def}}{206}$$

$$50 \rightarrow 194$$

$$\text{Def}(10, t) = 206 - \frac{206}{\lambda_0 - 40} (\lambda_0 - 40)$$

$$= 206 - 1/\lambda_0$$

$$\therefore P_d = \text{def} \times d$$

$$= 922.1 \times 201.8 \\ = 1848.34 \text{ kN} \quad (-R(t))$$

∴ The design compressive strength of the column will be 1848.34 kN.

Otherwise,

Let us consider it.

$$\text{i.e. about } \lambda = 2 \text{ and then } \phi_1 \\ P_d = 4 \phi_1 f_{ck} A_s \quad (\text{P-34, 2.1.1.1})$$

$$\therefore \lambda_{eq} = \frac{\lambda_1 / \lambda_{eq}}{\phi_1 + (\phi_1^2 - \lambda_1^2)^{1/2}}$$

$$\therefore \lambda_{eq} = \sqrt{\lambda_1^2 \left(\frac{f_{ck}}{n_e} \right)^2 / \lambda_1^2 \epsilon} \\ = \sqrt{\frac{212.5 \times \left(0.65 + 3.0 \times 10^{-3} \right)^2}{146.1}} = 0.474$$

The note: - when $\lambda < 0.3$, then the magnitude of λ to

$$\Phi_2 = 0.5 \left[1 + \alpha (\lambda_2 - 0.2) + \lambda_2^2 \right]$$

$$\alpha = 0.21 \quad (t=35, T=3)$$

$$\Phi_{22} = 0.5 \left[1 + \alpha \lambda_1 (0.2 - 0.2) + 0.2^2 \right]$$

$$\text{Ad} = \frac{\pi \times D / t \cdot l}{\pi \times D \cdot (D/2 - r)^2} \approx 2.27 \cdot 10^3 \text{ mm}^2$$

$$(P_A)_x = \text{Ad} \cdot \gamma (m)_x$$

$$= 0.921 \times 2.97 \cdot 2$$

$$= 8.092 \text{ N/mm}^2$$

(ii) about γ - γ axis and the η (b)

$$(P_A)_y = \text{Ad} \cdot \gamma (m)_y \approx 0.121 \times 2.97 \cdot 2 \approx 1.963 \cdot 10^{-3} \text{ N/mm}^2$$

$$\text{Ad} = \frac{d\gamma / \lambda_{yy}}{\frac{d\gamma}{D_y} \cdot (D_y^2 - \lambda_y^2) \cdot T} \approx 2.27 \cdot 10^3 \text{ mm}^2$$

$$\lambda_y = \sqrt{\lambda_y \left(\frac{R_y}{R_x} \right)^2 / k_{TF}} \approx 0.490 \text{ m}$$

$$\begin{aligned} \Phi_y &= 0.5 \left(1 + \alpha (\lambda_y - 0.2) + \lambda_y^2 \right) \\ &\approx 0.5 \left(1 + 0.34 (0.490 - 0.2) + 0.490^2 \right) \\ &\approx 0.669 \end{aligned}$$

$$\alpha = 0.34 \quad (t=35, T=3) \quad P_A = 1.963 \text{ N/mm}^2$$

Q) Design a wedge angle eccentricities fixed to carry a journal axial compression load of 100 kN. The length of travel is 2 m. Eccentricity is considered to be constant. Assume factor of safety as four times the safe working load. Use class 8 grade 400.

Sol:- For Fe 400,
 $f_u = 400 \text{ N/mm}^2$, $f_y = 300 \text{ N/mm}^2$.

$$f_{ek} = 1.125, f_{ed} = 1.1$$

For bolt of grade 4.6,

$$f_{ub} = 400 \text{ N/mm}^2$$

$$d_b = 12 \text{ mm}$$

$$d_s = 12 \text{ mm}$$

(el's. area modulus ratio $\lambda_e = 1.25$ and $\lambda_d = 1.0$).

From table 8.11)

$$f_{ek} = 23.7 \text{ N/mm}^2$$

Area required, $A = \frac{P}{f_{ek}}$

$$\therefore \frac{65 \text{ kN}}{23.7} = 2.77 \text{ dm}^2$$

From Steel table, 1st's profile 15A 70 x 70 x 6 mm.

Provided area = 40.6 mm^2 (F.S.)

$$e_{jk} = 13.6 \text{ mm}$$

Considering live and wind (F.W)

Effective length L = KxL

$$\therefore L = 1.5 \times 70 = 300 \text{ mm.}$$

Strength of walls

i) Crushing strength of walls

$$V_{crush} = \frac{f_{ck}}{f_s} (n_{v,crush} + n_{h,crush})$$

$$= \frac{400}{f_s} \left(1 + 0.25 + \frac{1.5 \times 10^2}{100} \right)$$

$$= 48.25 \text{ kN}$$

ii) Bearing Strength of walls

$$V_{bush} = \frac{f_b \cdot 0.5 \times K_{bush} \cdot T - \mu}{f_s}$$

$$= \frac{0.5 \times 1.5 \times 0.6 \times 100 \times 400}{100} = 90.0 \text{ kN}$$

∴ Strength of wall $\approx 48.25 \text{ kN}$.

No. of walls required for each construction

$$\approx \frac{60}{48.25} = 1.25 \approx 2 \text{ Nos.}$$

Therefore $2 \times 30 \text{ mm wide walls}$ for ensuring the load transmission of the panels.

Considering 20% end safety.

$$K_1 = 0.8, \quad K_2 = 1.25, \quad K_3 = 1.0, \quad (P = 40, \quad T = 12)$$

Then $\alpha = 0.49$ for case $(T=5, \quad T=1)$.

$$\lambda_{eff} = \frac{L/\pi_{eff}}{\sqrt{\frac{\pi^2 E}{300}}} = \frac{3000/12.6}{\sqrt{\frac{\pi^2 \times 20 \times 10^3}{300}}} = 0.482$$

$$\epsilon = \sqrt{\frac{dy}{dx}} = 1$$

$$\lambda_0 = \frac{b_i + b_o}{\sqrt{\frac{b^2}{250} + 91}} = \frac{70 + 70}{1 + \sqrt{\frac{7^2 + 7^2 + 91}{250}}} = 0.131$$

$$\lambda_c = \sqrt{0.2 + 0.3C \times 2.472^2 + 2.87 \times 0.131^2}$$

$$\phi = 0.5 + [1 + \alpha_1 (\lambda_c - 0.2) + \lambda_c^2] \\ = 0.1 + [1 + 0.49 + (0.131 - 0.2) + 0.0169] \\ = 0.2$$

$$f_{eq} = \frac{47 / f_{eq}}{\phi + (\beta^2 - \lambda_c^2)} = \frac{200 / 1.1}{2.0 + (0.20^2 - 0.131^2)} \\ = 63.02 \text{ N/mm}^2$$

design unit strength

$$P_u = \lambda_c \cdot f_{eq} = 0.131 \times 63.02 \\ = 8.29 \text{ kN} < 60 \text{ kN}$$

so design is not safe.

Now, let's provide 100 \times 100 \times 8 mm
provided area = 1008 mm^2 (1-4)

$$f_{eq} = 13.5 \text{ N/mm}^2$$

$$\lambda_{eq} = \frac{1/f_{eq}}{\sqrt{\frac{b^2}{250} + 91}} = \frac{1000 / 13.5}{1 + \sqrt{\frac{7^2 + 7^2 + 91}{250}}} = 0.5$$

(12)

$$\lambda_1 = \frac{(b_1 + b_2) / 2}{\sqrt{\frac{w^2 c}{\sigma v}}} \\ = \frac{(20 + 20) / 2}{\sqrt{\frac{1^2 + 0.0100}{0.1}}} = 0.098$$

$$\lambda_c = \sqrt{1.27 + 0.35 \times 2.4^2, 20 \times 0.098^2} \\ = 1.606$$

$$\phi = 0.5 \times \left[1 + \alpha \left(\lambda_c - 0.2 \right) + \lambda_c^2 \right]$$

$$= 0.5 \times \left[1 + 0.45 \left(1.606 - 0.2 \right) + 1.606^2 \right]$$

= 0.912

$$f_{ck} = \frac{f_t / Y_{t0}}{\phi + (\phi^2 - \lambda_c^2)^{1/2}} = \frac{250 / 1.1}{2.15 + (2.15^2 - 1.606^2)^{1/2}} \\ = 644.94 / \text{mm}^2$$

design compressive stress

$$P_d = A_e \times f_{ck}$$

$$= 1.52 \times 644.9$$

$$= 68.13 \text{ kN} > 65 \text{ kN}$$

So design is OK.

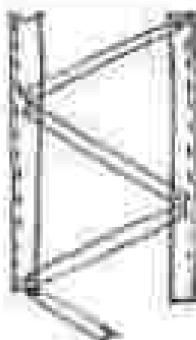
Loc of stiffened column

To reduce maximum stress for minimum section of column, without increasing the area of the section, a pair of struts are placed away from the principal axis which resist lateral system. The economy and lateral rigidity are also being obtained.

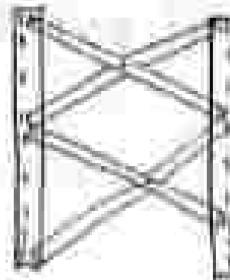
↳ ballooning -

using :-

Remember that webs and angles can resist shear loading. The object of providing lateral system is to keep the main members of the column away from pivoting ends.



(e.g. single angle system)

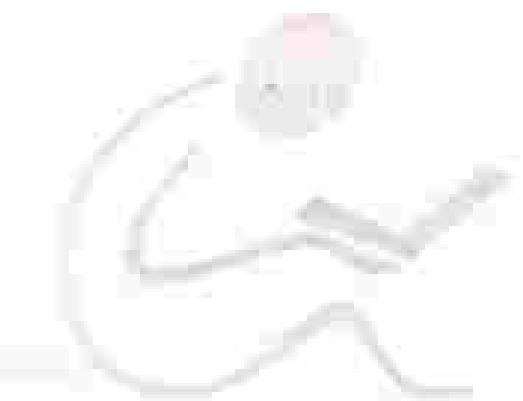
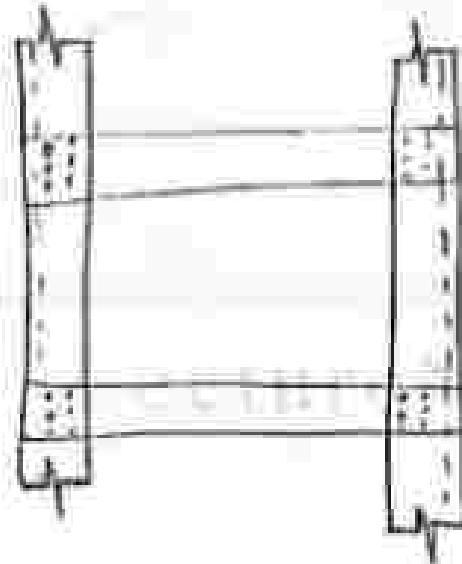


(e.g. multi-angle system)

(64)

Battens

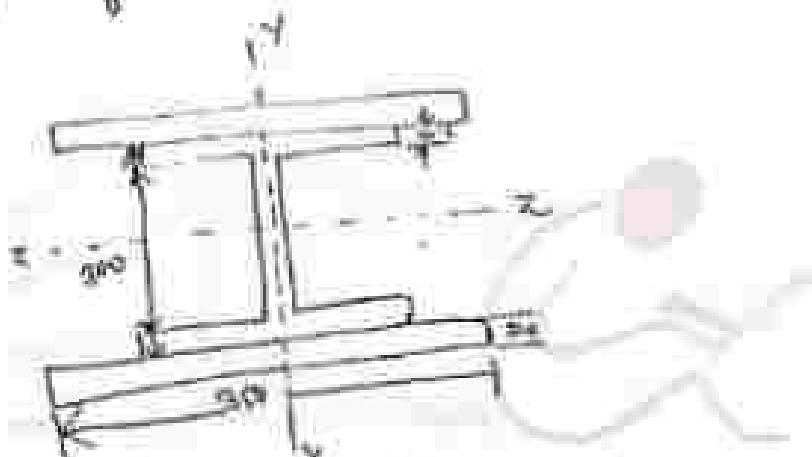
Instead of using an iron wire batten to keep members of column at required distance.



class. in

(fig: Battened column)

Calculate the compressive resistance of a compound column consisting of 16x16 size with one cover plate of 350×200 mm having a flange width of 5 mm. Assume that the bottom of column is fixed and top is rotation freely transversely free.



$$f_y = 350 \text{ N/mm}^2$$

$$L = 5000 \text{ mm}$$

from IS 800 : for 16x16 300 (P-14)

$$A = 74.95 \text{ mm}^2$$

$$I_{xx} = 19744.4 + 225 \times 10^4 \text{ mm}^4$$

$$G_{yy} = 24.93 \cdot 6 \times 10^3 \text{ mm}^4$$

Total area of built up section.

$$A = 74.95 + 2 \times (350 \times 200)$$

$$= 74.95 + 140000 \text{ mm}^2$$

$$T_{xx} = \text{int}(4t \cdot 3 \times 10^3) + \text{ext} \left[\frac{300 \times 30^3}{12} + 500 \times 30 \times \left(\frac{150 - 150}{3} \right)^2 \right] \\ = 665700 \cdot t \times 10^3 \text{ Nm} \\ = 44430 \cdot 2444.4 \text{ Nm}$$

$$T_{yy} = 2193 \cdot 6 \times 10^3 + 9 \times \frac{206300}{12} \\ = 16485 \cdot 76.8 \times 10^3 \text{ Nm}$$

Column will buckle about the axis of least moment resistance or least radius of gyration.

$$\therefore \sigma_{yy} = \sqrt{\frac{T_y}{I_y}} = \sqrt{\frac{16485 \cdot 76.8 \times 10^3}{21431}} \\ = 87.89 \text{ MPa}$$

Calculation for Design Stress

$$\sigma_y = \frac{K_c L}{r_y} = \frac{1.2 \times 5700}{87.89} \\ = 68.67 \text{ MPa}$$

From Table 19

Column will buckle about the axis of least moment resistance or least radius of gyration.

From 40.1

$\frac{\lambda}{\pi R_0 L_0}$	f_{cd}
0.00140	154.4

20 152 -

$$f_{cd}(154.4) = 168 - \frac{168 - 152}{30 - 20} \gamma (62.5 - 0) \\ = 152.4 \text{ N/mm}^2$$

$$P_d = 2192 \times 154.4 = 3312 \text{ kN (Ans.)}$$

Design of column base

(P-46, 7.4)

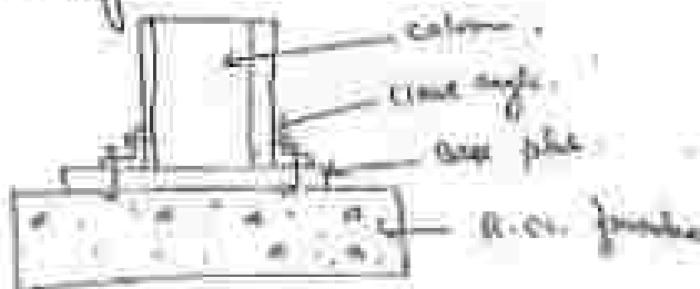
Column bases transmit the column load to the ground or foundation through bases. The column base affects the load on soil and also the stability of base factors on the foundation below. In addition, the column strength. There are two types of column bases connected.

(i) Stab base

(ii) Gravelled base

(iii) Slab base

These are used for columns carrying small loads. In this type, the column is directly connected to the base plate through steel angles. The load is transferred to the base plate through bearing.



(iv) Generalized base :-

For columns carrying heavy loads, generalized bases are used. In generalized bases, the column is connected to the base plate through girders. The load is transferred to the

~~long party through guest~~

69



Design of slab base

The design of slab base consists in finding the size and thickness of slab base.

Size of base slab

Step-1 Find the bearing strength of concrete which is given by $C = f_{ck} \times \text{unit width}$. In

Step-2 Therefore, area of base plate required

$$= \frac{P_u}{C \times f_{ck}}$$
 where P_u is factored load.

Step-3 Select the size of base plate. For economy as far as possible keep the projections of base to a minimum.

Thickness of base plate :-

Thickness of base plate

- (i) First intensity of pressure

$$\Delta = \frac{P_0}{\text{Area of base plate}}$$

- (ii) Minimum thickness required is given by (P.M.I. 74-2-1)

$$t_0 = \left[\frac{\pi \sigma (a^2 - 2ab)}{3\gamma} \right]^{1/3}$$

t_0 : Thickness of base plate

σ : Stress of Young

Design of quest base

- (i) Area of base plate = Product load / unit stress

- (ii) Minimum thickness of quest base

The thickness of quest plate is assumed as t_0 .

For ease of the quest angle is assumed such that the vertical leg is assumed to rotate and base can be provided.

In thickness of angle is kept approximately equal to the thickness of quest plate.

If quest plate

- (i) width of quest base is kept such that it will just project outside the quest angle and base.

Design = Area of plate
Width

- (ii) When the end loading of the column is considered for complete bearing on the base plate, 10% of the load is assumed to be transferred by the bearing of 10% of the loadings.

when the ends of the wires which are joint shall
be cut back by complete cutting, the cuttings
concerning them to the bare steel shall be disposed to prevent
all the parts to which they were attached.

(b) The thickness of the bare steel is controlled by thermal
strength of the welded joints.

Insulation Testing:-

Insulation testing shall begin at welding both the ground
wires to each other, except by the bare steel. These tests are
carried through the insulation by a test or by
other tests on the same wire appropriate level of insulation
resistance established in the normally dry condition under which
the test is performed, assuming an absence of sufficient to any moisture.

(c) Design a sheath resistance for a cable which can be compared
to an factor which compares with the one used for
the lightning conditions.

(d) Check the insulation to the bare steel by direct reading
of ohmmeter.

(e) Check the insulation to the bare steel by indirect
method; the losses due to the bare steel are
eliminated from reading.

(f) insulation design after these requirements?

For bare steel in normal condition of ground 1000 ft.

1000 ft = 100 ohms

1000 ft = 300 ohms

For 1000 ft of ground 1000 ft = 100 ohms

Bearing capacity of walls

$$= 0.45 \times f_u$$

$$= 0.45 \times 20$$

$$= 9 \text{ kN/mm}^2$$

$$f_{ck} = 61$$

$f_{ck} = 61$ (for soft soil)

For 1000 mm width $\Omega = 200 \text{ kN/mm}$. ($p = 14$)

$$t_1 = 11.6 \text{ mm}$$

$$t_2 = 10.1 \text{ mm}$$

$$h_1 = 300 \text{ mm}$$

$$h_2 = 675 \text{ mm}$$

Required area of slab base

$$A = \frac{1000 \times 10^3}{6 \cdot 9} = 166666.67 \text{ mm}^2$$

Let's provide a square base plate.

$$\text{Size of the base plate } l = 5 \times \sqrt{166666.67} \\ \approx 468.24 \text{ mm}$$

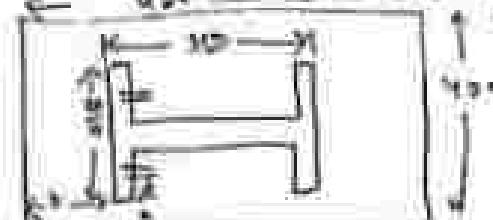
$$\approx 470 \text{ mm}$$

Let's provide base plate of $450 \times 450 \text{ mm}$.

The bearing pressure of walls

$$w = \frac{P}{A} = \frac{1000 \times 10^3}{450 \times 450}$$

$$w = 2.5 \text{ kN/mm}^2 < 9 \text{ kN/mm}^2$$



The greater proportion

$$a_1 = \frac{450 - 200}{2} + 20 \text{ mm}$$

The small proportion

$$b = \frac{450 - 200}{2} + 20 \text{ mm}$$

Thickness of each base

$$t_1 = \sqrt{\frac{0.1 \times (45^2 - 20^2) \times 100}{2}} \quad (P = 10, T = 0.1)$$

$$\sqrt{\frac{2.4 \times 10 \times (45^2 - 20^2) \times 100}{2400}}$$

$$\approx 20.39 \approx 20 \text{ mm} > 17.85 \text{ mm}$$

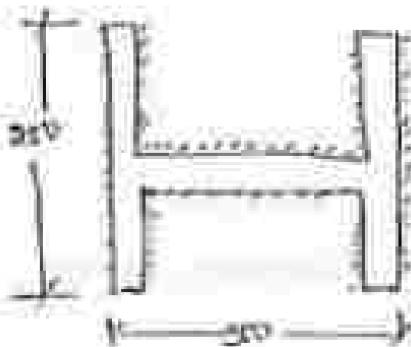
∴ Provide a base plate thickness of 20 mm in 2020.

iii) The load is transferred to the base plate by direct bearing. There is no bending moment. ∴ concentration of stress will not exist and will be uniform.
In order to keep the stress in tension, the total length of eccentricity of 20 mm may be provided increasing the distance from wall to the base plate.

iv) Column and base plate have to resist eccentric load for perfect bearing. Therefore, the load from the column will be transferred to the base plate through eccentric foundation. Length available for setting eccentric about center.

$$L_1 = 22000 + 20(450 - 20) + 2 \times (200 - 200 \times 0.1)$$

$$\approx 16200 \text{ mm}$$



We provide a new fixed width. Since cutting will not be made at the end, which of the flange section and column (25) will be unaffected at the end of each girder with length to get an effective length, that can be provided.

No. of total end notches = 10

Effective length = $1652.5 - 12.5$ (25)

$$(L) \quad \sim 1652.5 - 12.5 \times 10 \\ \sim 1497.5 \text{ mm}$$

(C.G. + 8 mm).

Effective thermal thickness = 0.125

$$= 0.125 + 0.6 = 0.75 \text{ mm}$$

Strength of the girder with

$$\text{Pmax} = \frac{f_u \times I_{eff} \times L}{\sqrt{3}} \\ \sqrt{3} = I_{eff}$$

$$= \frac{1497.5 \times 2.6 \times 10^6}{\sqrt{3} \times 0.75} = 152.5 \text{ kN} > 150.0 \text{ kN}$$

Assign to 10.

Since the base is subjected to only axial compressive load and there is no剪切, the base is not subjected to shear in any of the joints. Therefore, provide number 20 mm dia.

A clean white sand with rounded grains, fine-grained sand of fluvio-deltaic origin & suitable bottom grain size. The large rocks are all ^{gravel} coarse pebbles (10-24 mm) mostly of granite & boulders making the transition.

— From the 1900 census of Calif.

卷之二

by a man infatuated.

- 11 -

~~Resisting strength of concrete = $\frac{f_{c}}{\sqrt{1 + \frac{f_{c}}{f_{s}}}}$~~

• 111 •

- 1010-1/2000

לעומת מילון עברי-נוצרי.

卷之三

$\frac{d}{dt} \mathbf{v}_t = \mathbf{Q}_t \mathbf{v}_t$

— 17 —

— 30 —

training pitch (7) = 2.5 + 1
= 60 mm.

- 60 -

Aug. 20th (U) ~ 1944 ~ 39 min.

Leptospiral seroconversion

$$\text{Required area of base plate} A = \frac{1400 \times 10^3}{G \cdot S} = 231.45 \times 10^3 \text{ mm}^2$$

Well's provide the men with great pleasure on the long
of when when and the great angle
the 100% of the men.



Total width of base plate required

$$\approx 300 + 150 + 2 \times 10 = 610 \text{ mm} \\ \approx 612 \text{ mm.}$$

Proportion of base plate length to base angle bar

$$= \frac{612 - 60}{6} = 4 \text{ mm.}$$

Length of base plate = $\frac{l}{2}$

$$= \frac{612 - 60}{600} = 456.2 \text{ mm} \\ \approx 472 \text{ mm.}$$

Let's take 42 = base plate 630 mm in size.

Working tension of concrete $w = \frac{f}{4}$

$$= \frac{100 \text{ kg/mm}^2}{1500 \text{ kg/mm}^2} = 6.67 \text{ t/mm}^2 \\ < 6.75$$

Thickness of base plate :-

Let 't' be the thickness of base plate

The initial section of the base for better grip can be in section $t \times t$ as shown in Fig.

The length of base plate at critical position

$$(l) = 300 + 4 \times 10 = 340 \text{ mm.}$$

Let's consider thickness of base plate and span angle of working deck.

$$F_{\text{ext}} = m \cdot a = \frac{m \cdot v^2}{r}$$

$$\therefore \frac{C \cdot M \cdot 10^4}{r^2} = 26192.49 \text{ N/mm}^2$$

having empty support.

$$M_A = \frac{120 \cdot 200}{J_{\text{ext}}} \quad (\text{P.T.Y., I.G.I.D})$$

$$\therefore M_A = \frac{120}{7.1} \times \left(\frac{120^2}{C} \right)$$

$$= 42.44 \text{ t}^2 \quad - \text{Q}$$

Equation ④ & ⑤

$$120 \cdot 200 \text{ t}^2 + 26192.49$$

$$\Rightarrow t \approx 28.19 \text{ mm}$$

$$\text{radius of beam } R_1 = \frac{t}{2} = 14.095 \text{ mm}$$

$$= 28.19 - 14$$

$$= 14.09$$

$$\approx 14 \text{ mm} > R_2 = 10.6 \text{ mm}$$

Let's consider a beam which of length 1100 mm.

Bottled connection

Connection let's consider plate & flange, both will be in single shear and bearing.

$$V_{\text{flange}} = \frac{f_t}{2} (w_{\text{flange}} \cdot r_{\text{flange}}) \approx 0.021 \text{ kN}$$

$$V_{\text{plate}} = \frac{f_t (k_{\text{plate}} + k_{\text{flange}})}{2L} \cdot \frac{2 \cdot 100 \cdot 1100 \times 54.4 \times 10^{-3} \times 1000}{1.91} \approx 123 \text{ kN}$$

Let's consider the flange

$$\therefore \frac{C}{240} = 0.021$$

$$\therefore \frac{f_t}{40} = 0.021$$

$$\therefore \frac{f_t}{40} = 0.021 \approx 0.021$$

$$\therefore k_f = 1$$

$$\therefore k_f = 0.021$$

Strength of soil = 15.5 kN/m.

Assuming chain gathering and great resistance to have complete bearing. rest of the load will be removed to base diameter and rest of the load will be removed after consideration.

also of bolts required to connect the chain flange with gear plate

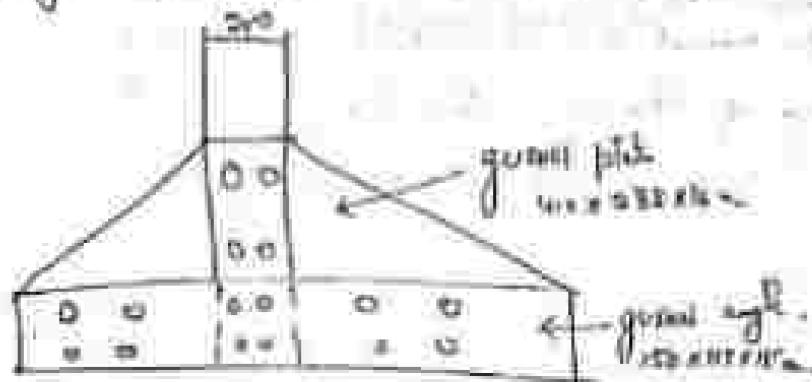
$$= \frac{0.5 \times 15.5}{0.5 - 0.1} = 15.5 \times 1.5 \text{ kN} =$$

This provides 30 mm dia bolt on each flange for gear plate. The no. of bolts required to connect the gear angle with gear plate will be same.

Length of gear plate = $120 + 2 \times 39 = 178 \text{ mm}$.

Length of gear plate + Length of base plate = 190 mm.

Provide gear plate thickness of 16 mm & width of 60.24 mm.



Design :-

A structural member subjected to pressure tends to resist a load. When pressure is applied to a structural member it resists load.

A beam supporting a load may either be called a simple beam or a beam of pure bending.

Design of beam :-

(i) Case 1 (Plastic) type :-

This section deals with plastic theory based on maximum capacity assumption. You assume all the material to be perfectly plastic.

(ii) Case 2 (Stress Analysis) type :-

This section deals with plastic moment of resistance. Here you assume plastic hinge formation causing plastic hinge mechanism due to local buckling.

(iii) Case 3 (Elastic Analysis) type :-

This one has sections in which the ultimate stress is considered to be safe yield stress, but cannot withstand the plastic moment of resistance due to local buckling.

Design of laterally supported beam :-

Design of laterally supported beam consist of identifying whether the type of condition of loading will result in the case of condition of eccentric and buckling. If the eccentric capacity is less than eccentricity, then buckling, lateral spreading and deflection etc.

Answers for:

- (Q) The shear force diagram on the beam can be written as
 $\text{Shear force} = \text{constant} + \text{variable}$ (Ans. $\text{Shear force} = C_1 + C_2 x$)
- (Q) The max bending moment 'M'_b, the shear force 'V' calculated
 from the above SFD points will differ due to change from
 A to B. Hence extra condition for the beam is required as
- $$V_B = M_B \times \frac{\partial}{\partial x} \quad (\text{P-65, 2-12})$$
- (Q) Looking at the value of shear force variation & bending
 moment varying slightly before and after point B, the required
 condition is $M_A = M_B$. Hence condition has been fixed.
- (Q) Ans.
- (Q) Consideration of section A is removed. From P-65, 2-12.)
- (Q) Consider the same bending (P-65, 2-12)
- (Q) The shear force is zero for shear capacity.
- (Q) The shear force is zero for design bending strength
- $$M_B = \frac{P \times L^2 \times f}{24} \quad (\text{P-65, 2-12})$$
- (Q) The shear force is zero for unloading
- $$F_B = \frac{(L_1 + L_2) M_B \times f}{J_{\text{eff}}} \quad (\text{P-65, 2-12}).$$

$$V < 0.649 \text{ (low shear) } (P=0.2, T=0.2)$$

$$V > 0.649 \text{ (high shear)}$$

(d) Check for adhesion:

$$\text{adhesive map} \quad (P=0, T=0)$$

Design for following simplified boundary conditions:-

(i) The surface will deform with the action of an action, the reaction force will increase with time from 'Jed' (\rightarrow) - otherwise the function fails.

(ii) The 'load' leading through 'act' may change from 'V' resulting from one shear stress from one adhered to an adhesive force.

(iii) A load shear action reaction force the shear is applied at

(iv) Acting on the side of your action magnitude a resistance reaction force having higher shear modulus more than required is obtained.

(v) Classification of shear is carried. ($P=0, T=0$)

(vi) Check for design safety margin

$$M_{\text{ad}} = \sqrt{\left(\frac{2^2 C_1}{(1-\nu)^2} \right) \left(0.3 + \frac{2^2 C_2 \nu}{(1-\nu)^2} \right)} \quad (P=0, T=0)$$

$$2_1 = \frac{2^2 C_1 \nu}{(1-\nu)^2} \quad (P=0.29)$$

$$2_2 = (1-\nu) \cdot 0.3 \cdot 2^2 C_2 \nu^2 \quad (P=0.29)$$

$$\text{for safe design} \\ M_{Q} = P_u + Z_p \cdot f_{u,i} > M_{u,r} (1 - 0.2Z_p, 2.25 + 0.2)$$

$$f_{u,i} = \frac{T_{u,i} \cdot f_y}{f_{u,r}}$$

$$q_{u,i} = \frac{1}{\sqrt{P_{u,i} + (P_{u,i}^2 - \lambda_{u,i})}} \leq 1$$

$$\Phi_{u,i} = 0.5 [1 + q_{u,i}(\lambda_{u,i} - 0.5) + \lambda_{u,i}^2]$$

$$T_{u,i} = \sqrt{\frac{P_{u,i} + Z_p \cdot f_y}{m_{u,i}}}$$

A simply supported beam just off the effective span is laterally supported at midspan. It carries a total uniformly distributed load, P_u . Design the appropriate section using the method of sections. Grade 50 steel.

Given: See the use grade of Steel.

$$f_y = 410 \text{ N/mm}^2 \quad f_{u,r} = 110$$

$$f_y = 355 \text{ N/mm}^2 \quad f_{u,r} = 115$$

$$\text{Span length} = W = 30 \text{ m}$$

$$\text{Factored load} = 1.5 \times 40$$

$$(W_u) = 60 \text{ kN/m} \times 30 \text{ m}$$

$$\text{Ultimate bending moment} = M_u = \frac{W_u L}{8} = \frac{60 \times 30}{8} = \frac{1800}{8} = 225 \text{ kNm}$$

$$\frac{60}{3} = 20 \text{ kN}$$

Plastic section modulus required

$$Z_f(\text{required}) = \frac{M_x Y_{eff}}{\sigma_y F_y} \quad (\text{P-43, Q-17})$$

$$= \frac{20 \times 10^3}{12 \times 200}, 125 \text{ cm}^3 \text{ mm}^2$$

From chart ICS 200 @ 100/2 n/mm $(T: 46, P: 137, \sigma: 1)$

$$t_f = 2.3 \text{ mm}, \quad Z_{eff} = 134.36 \text{ cm}^3 \text{ mm}^2$$

$$d_w = 5.4 \text{ mm}, \quad Z_{eff} = 134.36 \text{ cm}^3 \text{ mm}^2$$

$$b = 200 \text{ mm}, \quad T_{eff} = 12.96 \times 10^{-3} \text{ mm}^3 \quad (\text{Total width})$$

$$b_g = 150 \text{ mm}, \quad R_s = 7.6 \text{ mm} \quad (\text{From chart table (P-18, 1)})$$

$$R_s = 7.6 \text{ mm} \quad (\text{From chart table (P-18, 1)})$$

depth of web - of - h - 2 (t_f + R_s)

$$= 240 - 2 (7.6 + 2.3)$$

$$= 168.2 \text{ mm} \quad (\text{Total height})$$

Classification of column $(P=1)$

$$\varepsilon = \sqrt{\frac{200}{12}} \cdot \sqrt{\frac{200}{200}} = 1$$

$$\text{Half width of flange} = \frac{b_l}{2} = b = \frac{100}{2} = 50 \text{ mm}$$

$$\frac{b}{t_f} = \frac{50}{2.3}, 21.7 < 9.4$$

$$\frac{d}{t_w} = \frac{168.2}{5.4}, 31.1 < 94$$

\therefore The section is Plastic.

(E3)

check for axial bending :- (P-43, Q-2.1.3)

$$\text{Since } \frac{d}{t_{\text{min}}} = 30 > 6 \Rightarrow$$

Thus $\frac{d}{t_{\text{min}}} < 6$, axial bending check will not be required.

check for shear capacity :- (P-44, Q-1)

$$\text{Required force} = 30 \text{ kN}$$

$$V_u = \frac{\text{Applied}}{V_{\text{allow}}} = \frac{30 \times 10^3 \text{ N/mm}^2}{150 \text{ N/mm}^2} = 200 \text{ kN} > 30 \text{ kN}$$

$V_u = V_{\text{allow}}$ for not reduced value (Q-1.1)

$$= 200 \text{ kN}$$

$V_u > V$. So design is OK.

check for aligned bending strength :- (P-43)

$$M_u = \frac{M_{\text{allow}} t_{\text{min}}}{T_u} \quad (\text{P-43, Q-2.1.3})$$

$$= 1.7 \frac{100 \times 10^3 \text{ Nmm}}{100} = 17.295 \text{ kNm}$$

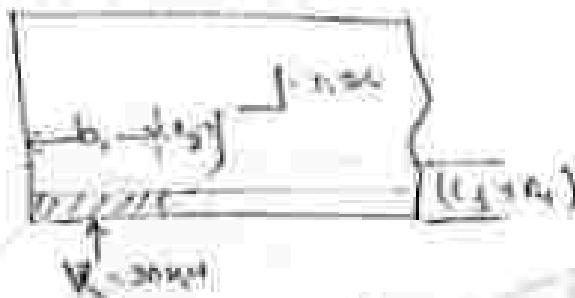
$$M_d \leq 1.2 T_u \frac{17}{T_u} = 1.2 \times 100 \times \frac{10^3 \times 10^3}{100} = 120 \text{ kNm} > 30 \text{ kNm}$$

$M_d > M_u$
So design is OK.

~~check for safe bearing capacity~~ (84)

$$F_u = \frac{(b + t_0)}{T_{0e}} t_0 f_y = 147.59 \text{ kN/mm}$$

allowing safety factor



allowing safety bearing stress $\lambda_s = 75$ mm.

t_0 = design thickness of flange.

$$\therefore \pi r^2 (t_0 R_0) = 25 (7.314)^2$$

42 mm.

Lecturing position in

$F_u > F$

So design is OK.

check for width/base ratio (P-17, P-2-1.2)

$$0.6 V_d = 0.6 \times 140 \times 75 \\ \approx 45.00 \text{ kN/mm.}$$

$\text{Since } V < 0.6 V_d.$

so it is safe design.

Check for deflection (P-31, T=6)

$$\text{Permissible deflection} = \frac{\text{Span}}{300} = \frac{4.8\text{m}}{300}$$

$$\approx 13.33\text{mm.}$$

$$\text{Actual deflection} = \frac{5}{384q} = \frac{5 \times 10^3}{48}$$

$$\Rightarrow \frac{5}{384q} = \frac{\pi(64) \times 1^2}{EI}$$

$$\Rightarrow \frac{5}{384q} = \frac{4 \times 10^{-3} \times (4710)^2}{2 \times 10^2 \times 102.6 \times 10^3}$$

$$\Rightarrow q \leq 12.33\text{mm.}$$

So design is OK.

Q Design a laterally unbraced beam for the following data

$$\text{Effective Span} = 4\text{m}$$

$$\text{Max bending moment} = 550 \text{ Nm}$$

$$\text{Max shear force} = 200\text{N/mm} \quad q = 96.953 \times 10^3 \text{ N/mm}^2$$

$$\text{Mod of grade: Fe 410.} \quad (\text{Cross section})$$

Ans: For Fe 410 grade of S400.

$$I_y = 300 \text{ mm}^4$$

$$f_y = 410 \text{ N/mm}^2$$

$$T_{bd} = 1.1$$

$$T_{ad} = 1.15$$

PLASTIC ACTION THEOREM requires

$$\gamma_{p2} (\text{Required}) = \frac{1.5 \times M_d \times l_{pl}}{I_y}$$
$$= \frac{1.5 \times 150 \times 10^3 \times 11}{6.07}$$
$$= 304.5 \times 10^3 \text{ mm}^3$$

Given related values are $E = 200 \text{ GPa}$, $T = 75^\circ\text{C}$.

$$l_1 = 600 \text{ mm}$$

$$E_{pl} = 0.12 \times 200 \times 10^3 \text{ GPa}$$

$$l_2 = 250 \text{ mm}$$

$$E_{pl} = 0.05 \times 200 \times 10^3 \text{ GPa}$$

$$l_3 = l_4 = 250 \text{ mm}$$

$$l_{pl} = 150 \text{ mm}$$

From yield charts ($\tau = 14$)

$$E_{pl} = E_0 = 200 \text{ GPa}$$

$$T_2 = 91.7 \times 10^3 \text{ mm}^3$$

$$I_y = 200 \times 10^3 \text{ mm}^4$$

$$d = 4 - \sqrt{4(4R_1)}$$

$$= 6.0 - (114 + 23)$$

$$= 518.4 \text{ mm}$$

Section classification

$$c = \sqrt{\frac{320}{3y}} < 1$$

$$\text{Reinforced girder } b = \frac{b_1}{2} + \frac{240}{2} = 108 \text{ mm}.$$

$$\frac{b}{4} = 27 \text{ mm} < 34$$

$$\frac{d}{4} = 129 \text{ mm} < 34$$

∴ Section is Plastic

Check for design bending strength

$$M_{pl} = \sqrt{\frac{\pi^2 EI_y}{(L_{pl})^2} \left(L_{pl}^4 + \frac{\pi^2 EI_0}{(L_{pl})^2} \right)}$$

$$E = 2 \times 10^3 \text{ N/mm}^2$$

$G = 0.75 \times 10^3 \text{ N/mm}^2$ (Shear modulus)

$\Delta L_1 = 10 \text{ mm}$

$$\Delta L_1 = \frac{M_1 l_1^3}{3} \quad (\text{Eq. 1.19})$$

$$= 2 \times \frac{210 \times 70 \times 10^3}{3} \times (0.12 - 0.05) \times 10^3 \\ = 4.2 \times 10^6 \text{ mm}^3$$

$$240 = (k - k_1)(M_1 l_1 \frac{l_1^2}{3}) = (k - 90) \pi \times 2651 \times 10^3 \times 10^3 \text{ N/m}^2$$

$$k_1 = \frac{340}{l_1 + l_2} = 340$$

$$\text{Hence } k_{\text{eff}} = k_1$$

L_2 = Distance w.r.t shear center of two flanges of the S.I.

$$= 670 - \frac{210}{3} - \frac{20 \times 2}{3} = 579.3 \text{ mm}$$

$$M_{\text{eff}} = \sqrt{\frac{\pi^2 \times 210^3 \times 2651 \times 10^3}{4000^2}} = \sqrt{36.92 \times 10^6 \times 10^3} = \frac{6075 \times 10^3}{4000}$$

$$= 15.2 \text{ kNm}$$

M_{eff} Bridge bending moment:

$$M_{\text{eff}} = P_1 \times L_2 \times h_2 \quad (P_1 = 17.9 \text{ kN}) = 597 \text{ kNm}$$

$$I_{\text{eff}} = \frac{A_{\text{eff}} h^3}{3} = 17.9 \times 0.6 \text{ m}^4$$

$$\sigma_{\text{eff}} = \frac{\sigma_{\text{m}}}{\sqrt{(\sigma_{\text{m}})^2 - (\sigma_{\text{eff}})^2}} \approx 0.7983$$

$$\Phi_{\text{eff}} = 0.5 \left[1 + K \left(\alpha \lambda_{\text{eff}} - \alpha_0 \right) \sqrt{\lambda_{\text{eff}}} \right] \approx 0.966$$

$K = 0.0002$ ($\alpha = 0.21$)

$$M_1 = \sqrt{\frac{\beta_{\text{eff}} Z_{\text{eff}}^2 \gamma}{4 \pi n}} = \frac{1.1366 \times 10^3 \times 210}{11 \sqrt{3.3} \times 10^2}$$

$\gamma = 0.878$

Ans. Not OK.

So design is OK.

Check for shear capacity :-

Design shear force $V = 200 \text{ kN}$.

Design shear strength of the section

$V_u = \frac{f_y A_s}{25 f_{ck}}$

$f_y = 300 \text{ N/mm}^2$

$$= \frac{300 \times 12 \times 210}{25 \times 300} = \frac{60 \times 12 \times 210}{\sqrt{3} \times 10^3} = 944.38 \text{ kN}$$

$V_d > V_u$. So design is OK.

Check for web buckling

$\frac{d}{t_w} < C_f c$ \Rightarrow web for web buckling is not required.

$$\text{If } \frac{d}{dx} > 6.7 \text{ C.}$$

so how to check capacity of detector.

Capacity of detector = $\lambda_{\text{eff}} \cdot t_{\text{eff}} > V$

$$A_p = (5 \times 10) \text{ cm}^2 \approx (10 \times 10) \text{ cm}^2 \approx 100 \text{ cm}^2,$$

$t_{\text{eff}} = \text{detection length} = 100 \text{ mm. (assumed)}$,

$$t_{\text{eff}} = \frac{h}{2} = \frac{100}{2} = 50 \text{ mm.}$$

$$\lambda_{\text{eff}} = \frac{\lambda_0}{n} = \frac{365 \cdot 10^{-9}}{1.375} = 269 \cdot 10^{-9} \text{ m.}$$

$A_p = \text{Efficiency length of detector} = 50 \text{ mm}$

$$= 0.7 \times 10^{-12} \text{ A}$$

$$= 3.63 \cdot 10^{-13} \text{ A m.}$$

$$n = \sqrt{\frac{A_p}{A_p \cdot t_{\text{eff}}}} = \sqrt{\frac{100 \cdot 10^{-4}}{100 \cdot 10^{-3}}} = 3.16$$

$$\lambda_{\text{eff}} = \frac{100 \cdot 10^{-9}}{1.375} \approx 144 \cdot 10^{-9} \text{ m. Thus, in}$$

$$\text{length of } A_p = 100 \cdot 10^{-3} = 100 \text{ mm.}$$

Assuming $\text{Cross Sec. } \text{for } \lambda = 144 \cdot 10^{-9} \text{ m. } 10 \text{ fm}^2$

$$\text{Rate} = \frac{\lambda}{100} \cdot \frac{100}{100} = \frac{100}{100} \cdot \frac{100}{100} =$$

$$\frac{100}{100} (\text{rate}) : \text{rate} = \frac{100 - 269 \cdot 10^{-12}}{100 + 100} \left(100 \cdot 10^{-12} - 100 \right)$$

$$\approx 100 \cdot 10^{-12} \text{ A/m}^2 \text{--}$$

capacity of detector = $100 \cdot 10^{-12} \text{ A/m}^2$

$$\therefore 100 \cdot 10^{-12} > 100 \text{ A/m}^2$$

check for wr. bending

$$f_w = \frac{(n + n_1) f_{w,typ}}{f_{w,c}}$$

$$b = 100 \text{ mm}$$

$$n_1 = 2.5(t_3 + R_1) = 102 \text{ mm}$$

$$f_w = 550 \text{ N/mm}^2 > V$$

So design is OK

(10)

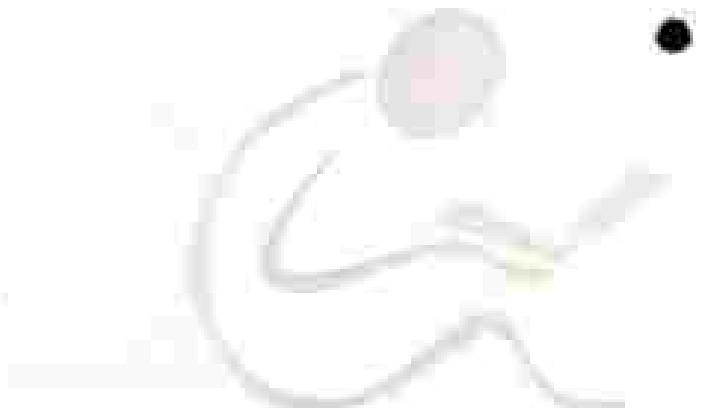
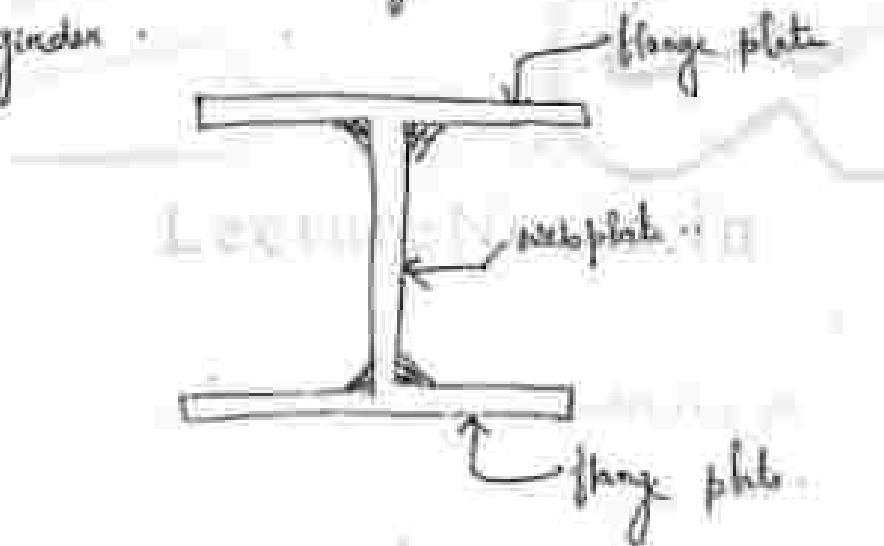


Plate Girder

When span and load increase, the available rolled section may be insufficient, case often strengthening with extra plates. Such girder is known as plate girder.

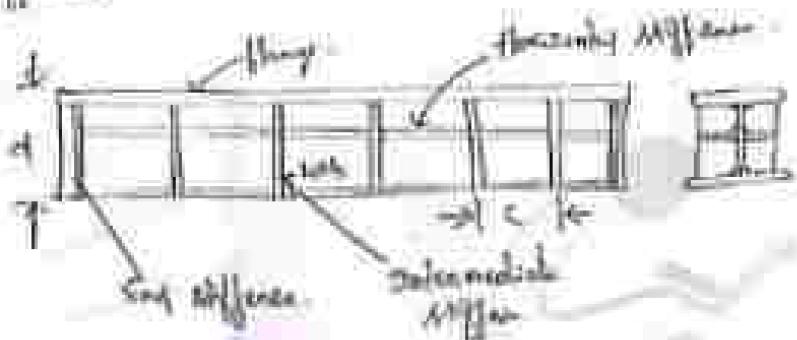
Diagram on next station one of the remedies is to go for a built up T-beam with two flanges held together by a central plate of required depth. The depth of such T-beams may vary from 1.5 m to 5 m. This type of T-beams are known as plate girder.



Elements of plate girder.

Following are the elements of a typical girder:-

- ① Web.
- ② Flange.
- ③ stiffener.



(i) Web:

Area of required width and thickness are provided to carry shear force at required distance.

(ii) stiffener :-

(iii) Flange:

Plates of required width and thickness are provided to resist bending moment acting on the beam by applying compressive force in one flange and tensile force in another flange.

(iv) Edge:

Stiffeners are provided to safeguard the web against local buckling failure. The stiffeners passing along the flanges

- (a) Transverse (vertical) differences
- (b) Longitudinal (horizontal) "

(c) Intermediate differences are of two types.

(i) Bearing differences -

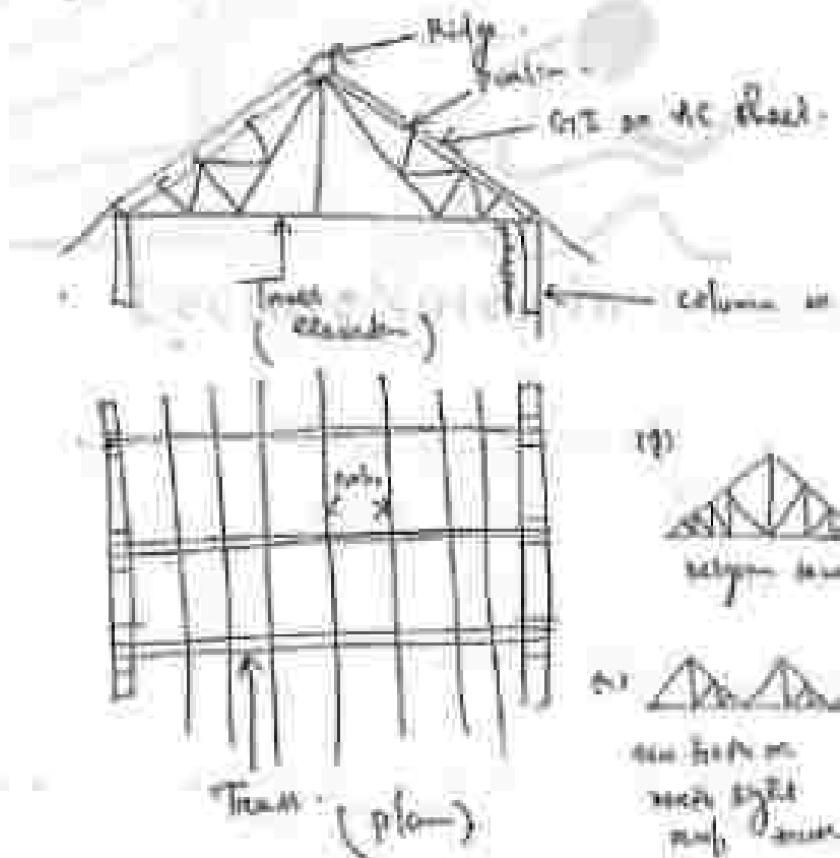
(ii) Intermediate differences -

The bearing differences are provided to distribute the load more to one support. At the end certain portion of web of beam acts as a compression member and hence there is possibility of cracking of web. Hence web needs differences to transfer the load to the support. If concentrated loads are acting on the girder, intermediate bearing differences are required.

Ridge Trusses

(2)

Large open free areas are required for audience viewing halls, auditoriums etc. To get such volumes free and out of the economy roof carrying system is to provide a set of steel roof trusses, interconnected by purlins which in turn support G.I.C (Galvanized Iron) or AC (Asbestos Cement) sheets. The roof trusses are supported on walls or a series of columns.

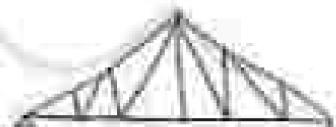


Types of roof trusses :-

- On the basis of structural behaviour, roof trusses can be classified as simple roof trusses supported over many / concrete walls / columns and rigid frames.
- A roof frame may also be called a plane frame or space frame.
- In a plane frame the members make up the component members like in the same plane whereas in a space frame the component members are enclosed in a dimension in space and hence they can act in any direction.



(a) Simple roof truss.



(b) Plane frame.



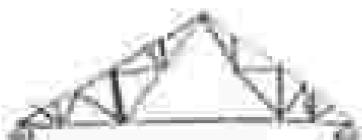
(c) Flying buttress frame.



(d) Fan truss.



(e) Fink frame truss.



(f) Compound roof frame.